

Engineering Disasters: Learning from Failure

N. Chawla

School of Materials
Fulton School of Engineering
Arizona State University
Tempe, AZ 85287-8706

nchawla@asu.edu

<http://enpub.fulton.asu.edu/chawla/Home.htm>
(480) 965-2402 (phone) (480) 727-9321 (fax)

Submitted to:

TMS Education Resource Awards
November 2007

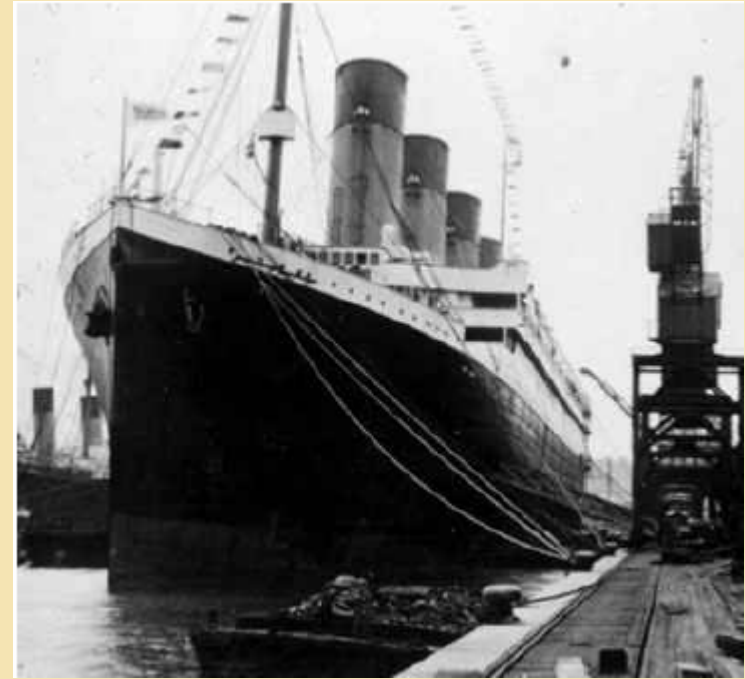


Engineering Disasters: Learning from Failure

- Engineering disasters have resulted in loss of life, injuries, and billions of dollars in damage.
- Primary causes for engineering disasters:
 - Design flaws
 - Material failures
 - Extreme conditions or environments (not necessarily preventable)
 - Some combinations of the reasons above.
- Three major disasters:
 - Sinking of the ship Titanic
 - Collapse of the World Trade Center buildings
 - Explosion of the Space Shuttle Columbia

Chronology of Events Leading to Sinking of the *Titanic*

- Titanic began its maiden voyage to New York at noon on April 10, 1912, from Southampton, England.
- On night of April 14, at 11:40 p.m., crew sighted an iceberg immediately ahead of ship.
- In about 40 seconds it collided with an iceberg estimated to have a gross weight of 150,000-300,000 tons.
- Iceberg struck the *Titanic* near bow and raked side of ship's hull damaging hull plates and popping rivets,
- At 2:20 a.m., April 15, 1912, *Titanic* sank within two hours and 40 minutes, with the loss of more than 1,500 lives.



Why did the *Titanic* sink?

- Theory 1

Multiple rivet failures upon collision with iceberg.

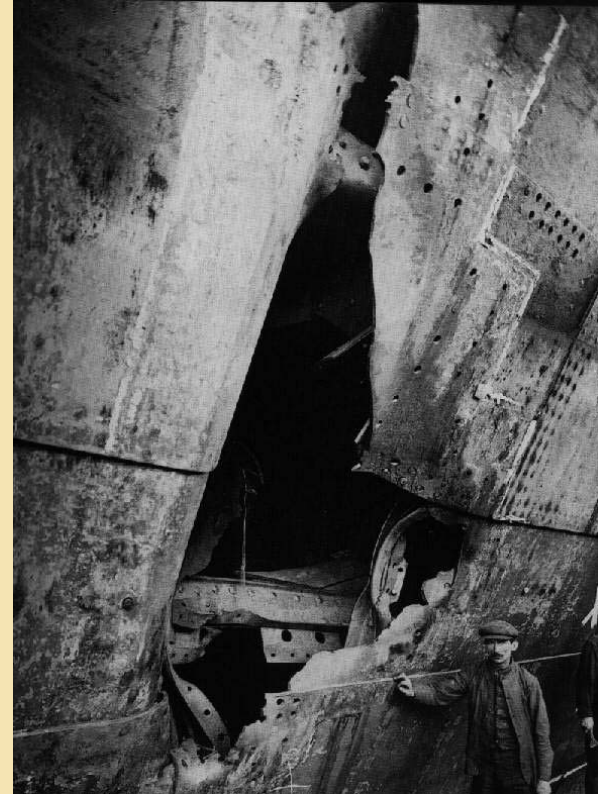
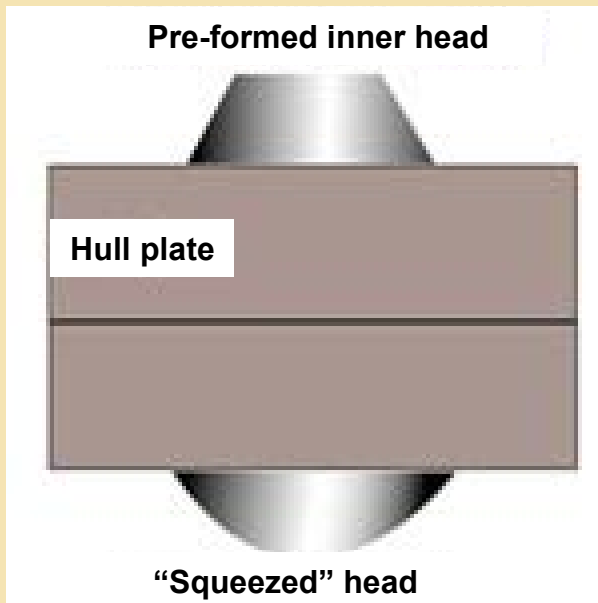
- Theory 2

Failure of the steel hull upon collision with iceberg.

Remains of the *Titanic*



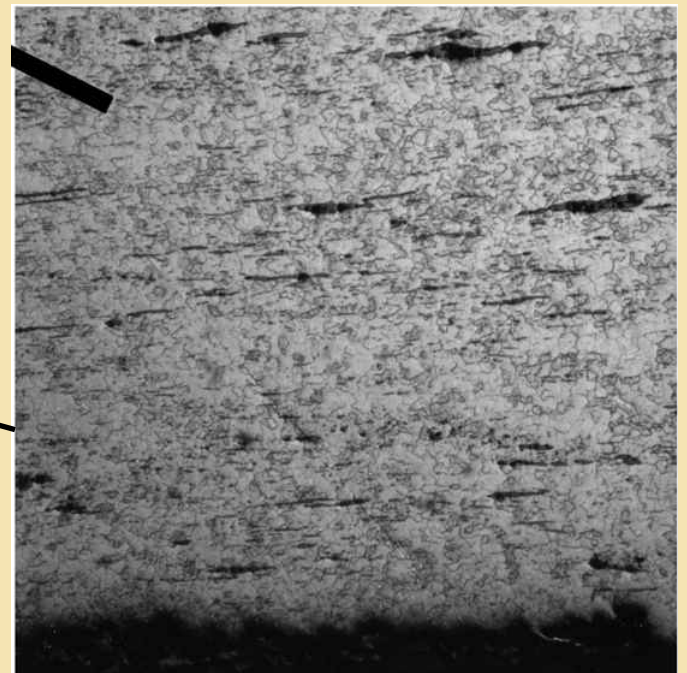
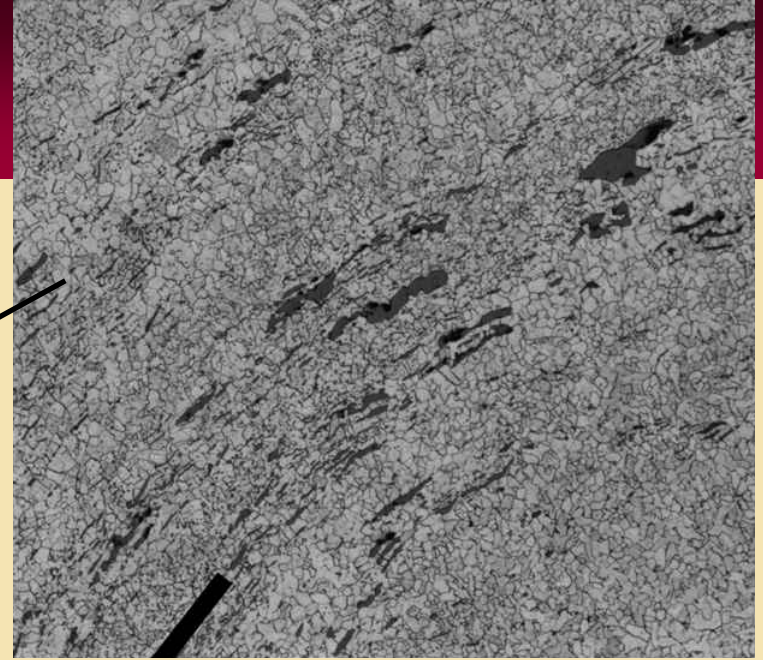
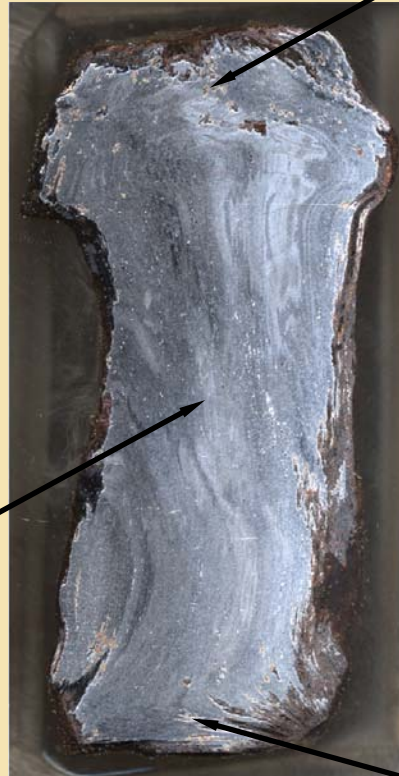
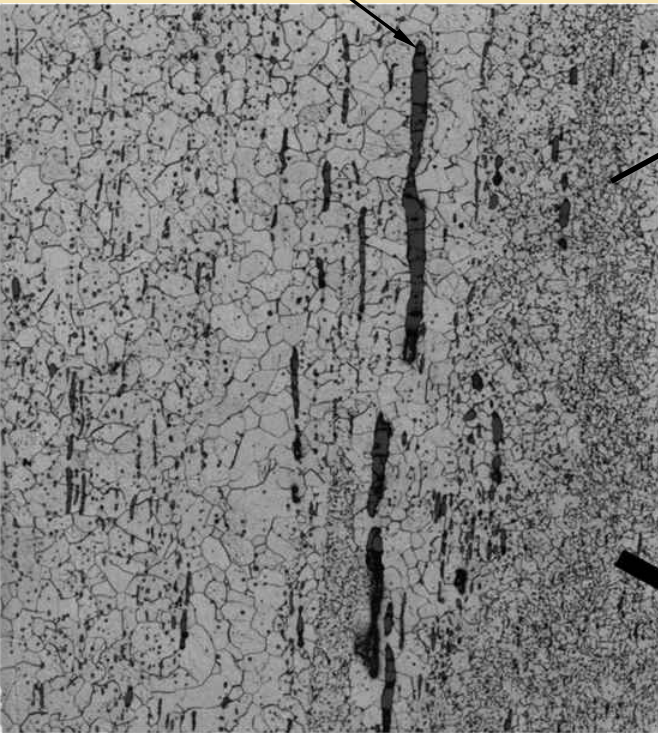
Theory 1: Multiple wrought-Fe rivet failures upon collision with iceberg.



Hull of the *Olympic*, *Titanic's* sister ship after a collision in 1911.

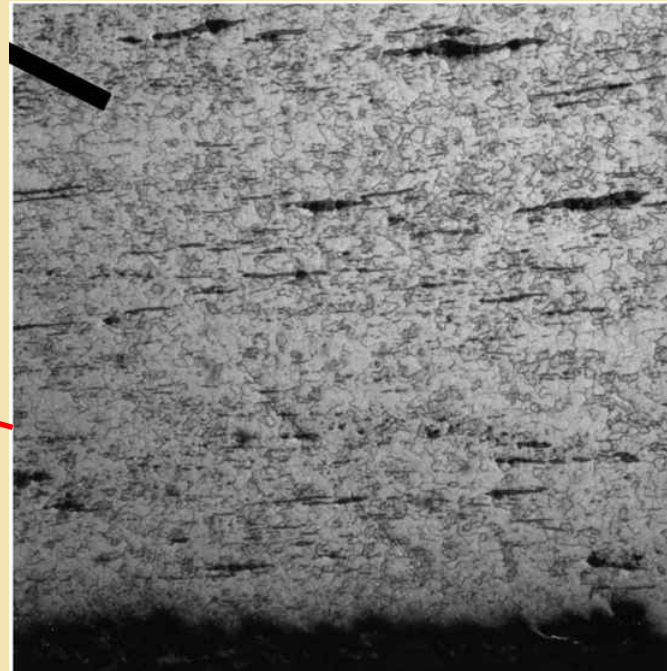
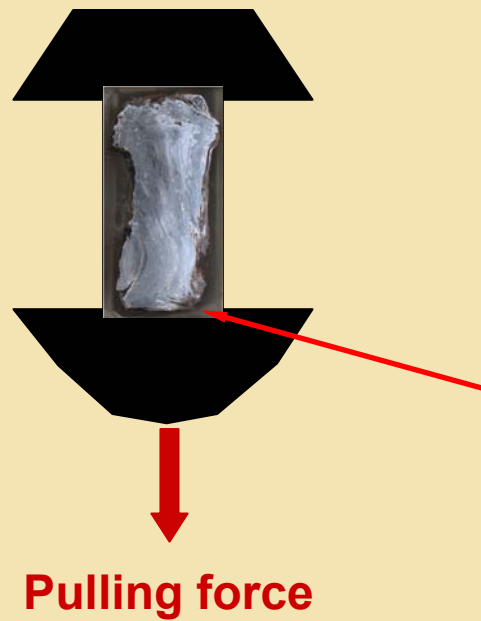
Microstructure of Titanic Rivet

Fe-silicate "stringers"



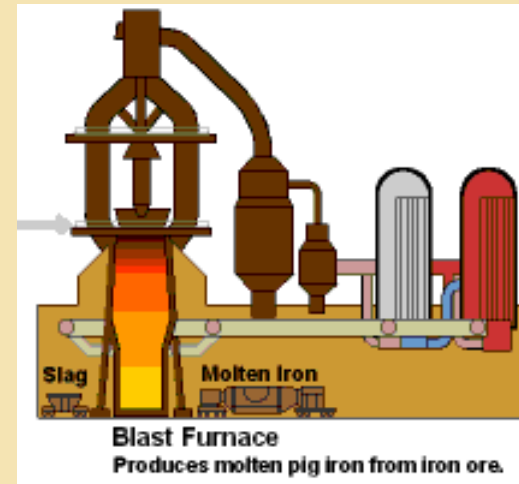
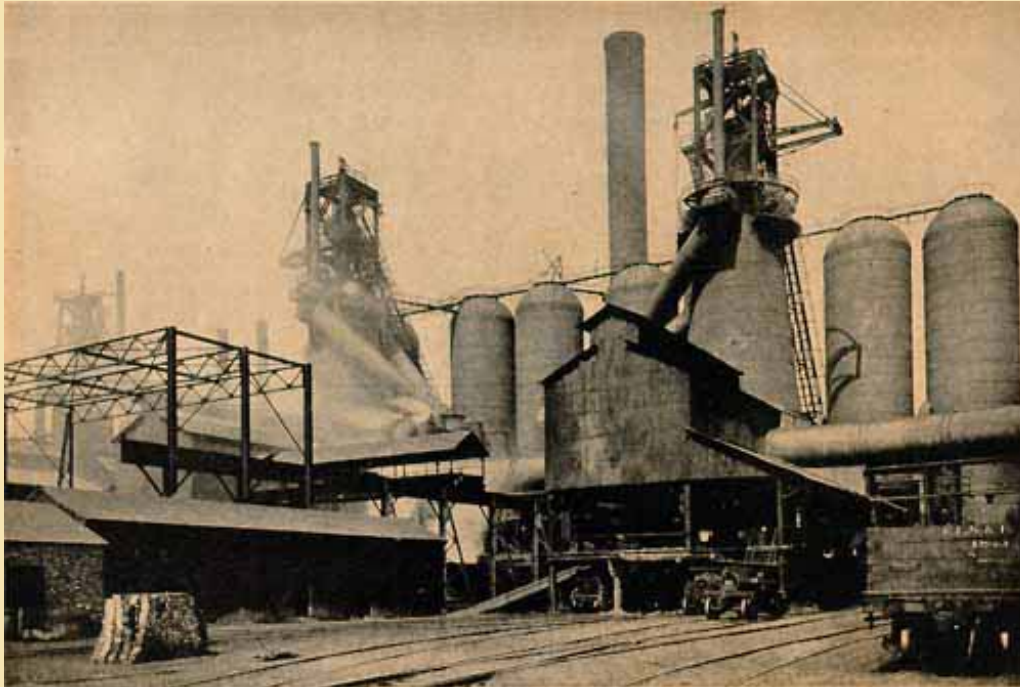
Microstructure of Titanic Rivet

- Orientation of Fe-silicate stringers is perpendicular to loading axis at the end of the rivet
 - Much lower strength and inferior resistance to crack propagation



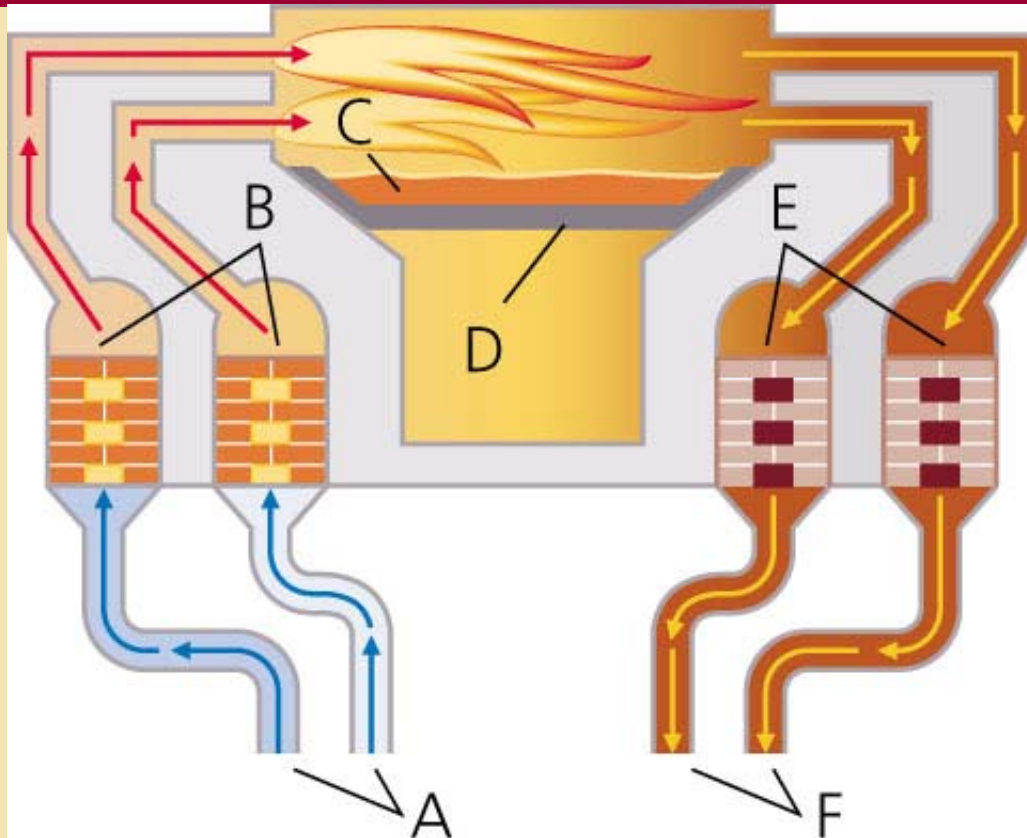
Theory 2: Hull Fracture the low carbon steel hull upon collision with iceberg.

Blast Furnace Process



- Iron ore, coke, and limestone are raw materials which are charged at the top of the blast furnace.
- Molten “pig iron” and slag are collected at the bottom and are tapped out at intervals.

Open-Hearth Furnace Process



- A. Gas and air enter**
- B. Pre-heated chamber**
- C. Molten pig iron**
- D. Hearth**
- E. Heating chamber (cold)**
- F. Gas and air exit**

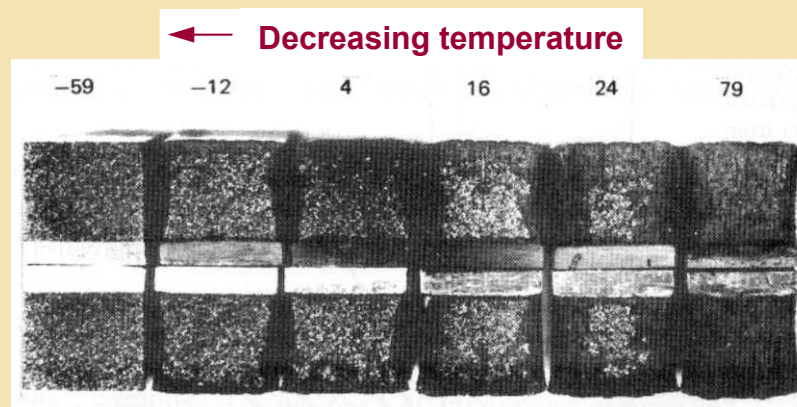
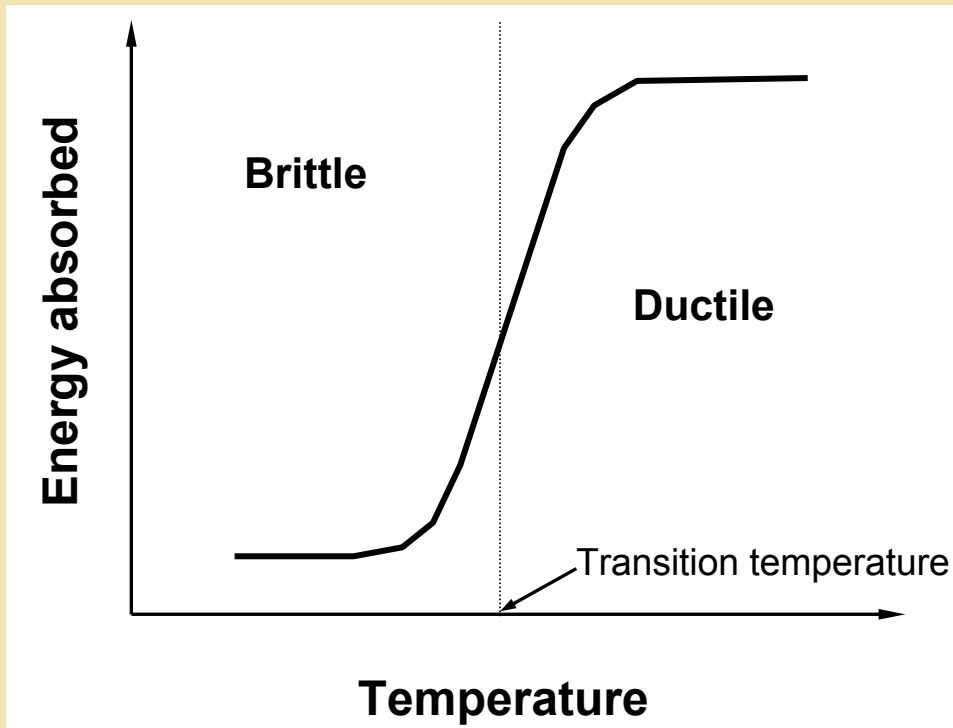
- In acid open-hearth steel process, an acid material, silica, is used as the furnace lining.
- Pig iron (92% Fe and about 3.5% C) is charged in. Impurities, including carbon, are oxidized and float out of the iron into the slag.
- Siliceous refractory material in the lining will not react with P or S so the content of these elements in the steel will be extremely high.

Comparison of Chemical Composition of *Titanic* Hull Steel vs. Modern Steels

Material	C	Mn	P	S	Si	Cu	O	N	Mn-to-S Ratio
<i>Titanic</i> Hull Plate	0.21	0.47	0.045	0.069	0.017	0.024	0.013	0.0035	6.8:1
ASTM 36	0.20	0.55	0.012	0.037	0.007	0.01	0.079	0.0032	14.9:1

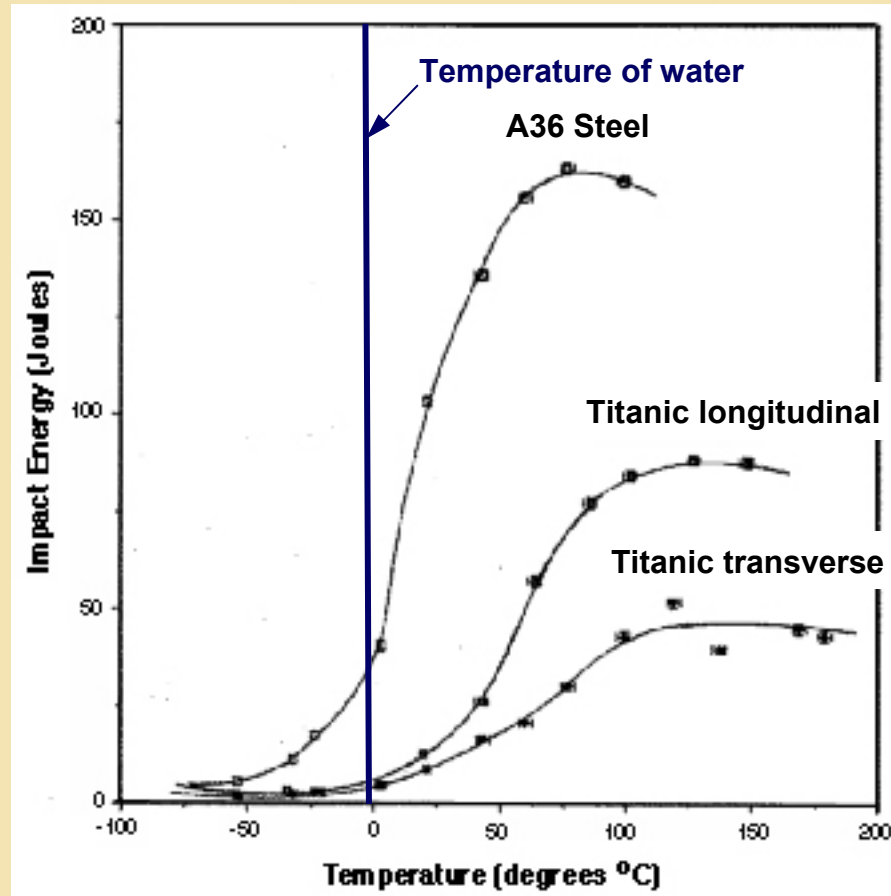
- Mn-to-S ratio is lower and P content slightly higher in *Titanic* Hull Plate than in modern steels of similar composition.
- Higher S and P amounts can be attributed to acid furnace lining used in open-heart furnaces of that time period.

Ductile-to-Brittle Transition



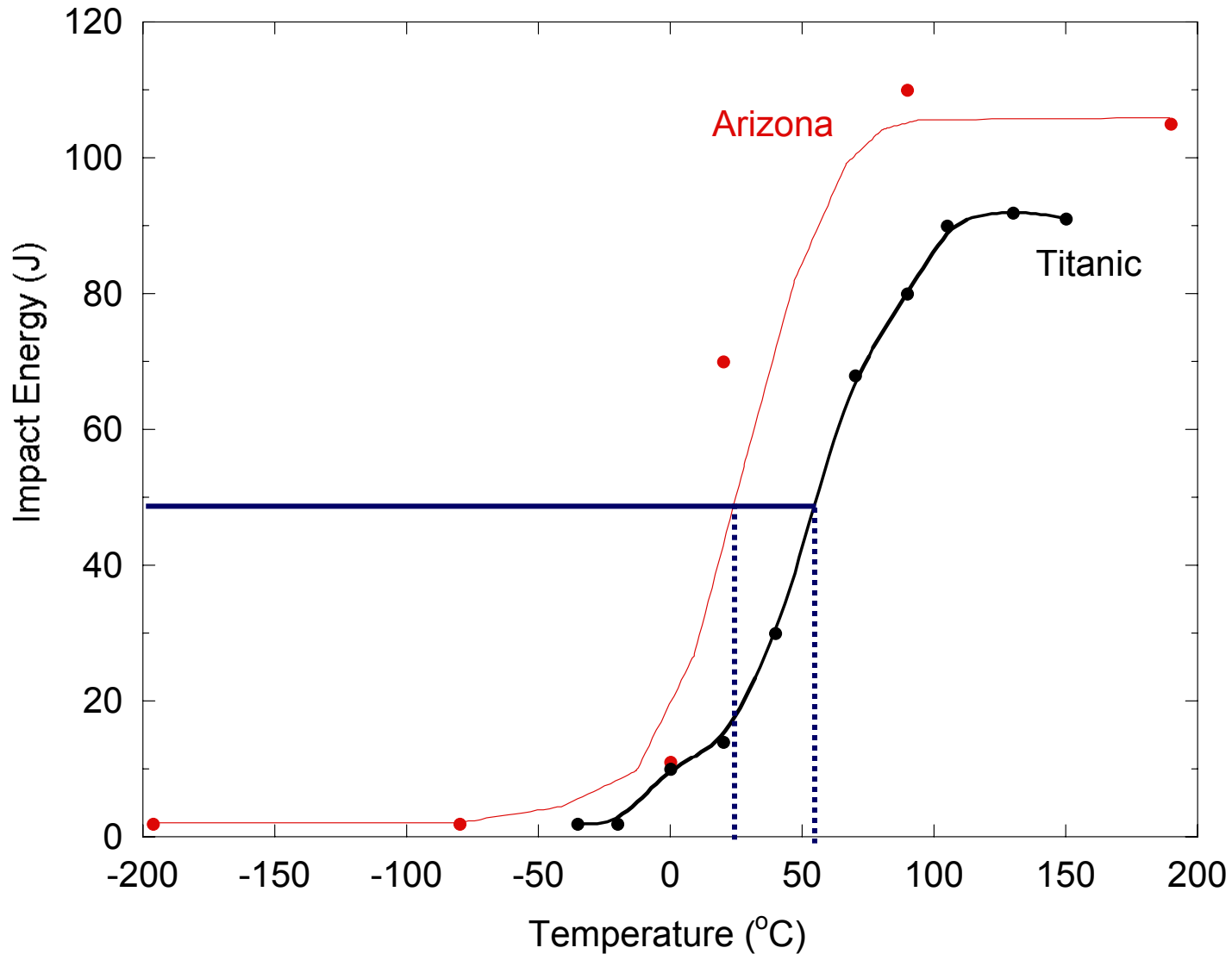
Fracture is more "brittle" with decreasing temperature

Impact Energy versus Temperature



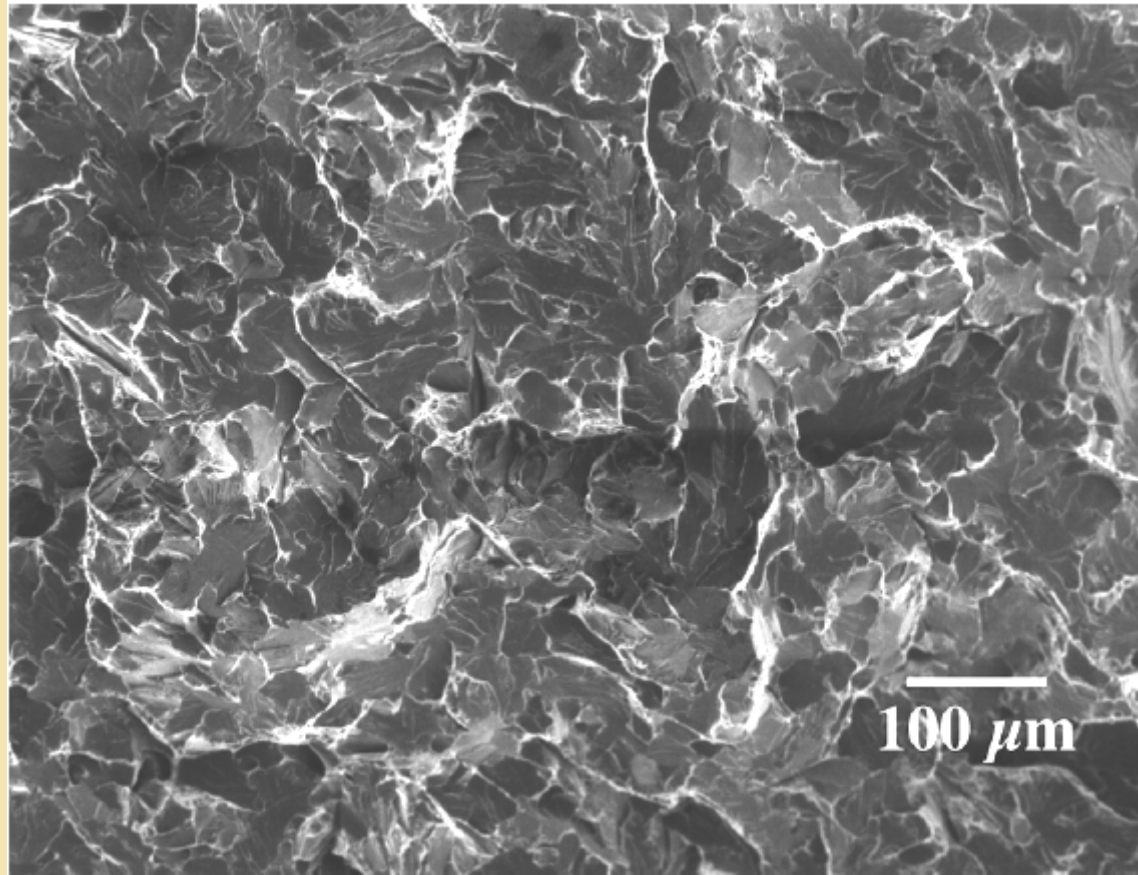
Temperature of the water was -2°C !!

Impact Toughness of Steel in *Titanic* vs. *U.S.S. Arizona*



Steel of U.S.S. Arizona had lower DBTT!

Fracture Surface of Titanic Steel (Longitudinal) Impacted at 0°C



Characteristic brittle fracture is observed

Lessons Learned from the Sinking of the *Titanic*

- Mn significantly decreases DBTT.
 - Titanic steel was low in Mn, most of which likely combined with S to form MnS
- Finer grain size improves toughness and decreases DBTT. Fine grain structure is achieved by deoxidation practice.
 - Titanic steel appears to be only partially deoxidized (note high oxygen content)

These factors appear to have contributed to a higher DBTT in Titanic Steel, which made it extremely brittle at the water Temperature (-2°C)

Modern Steels Have Much Higher Toughness and Lower Ductile-To-Brittle Transition Temperature

The World Trade Center Buildings

Height: 1,368 and 1,362 feet (417 and 415 meters)

Owners: Port Authority of New York and New Jersey

Architect: Minoru Yamasaki

Ground Breaking: August 5, 1966

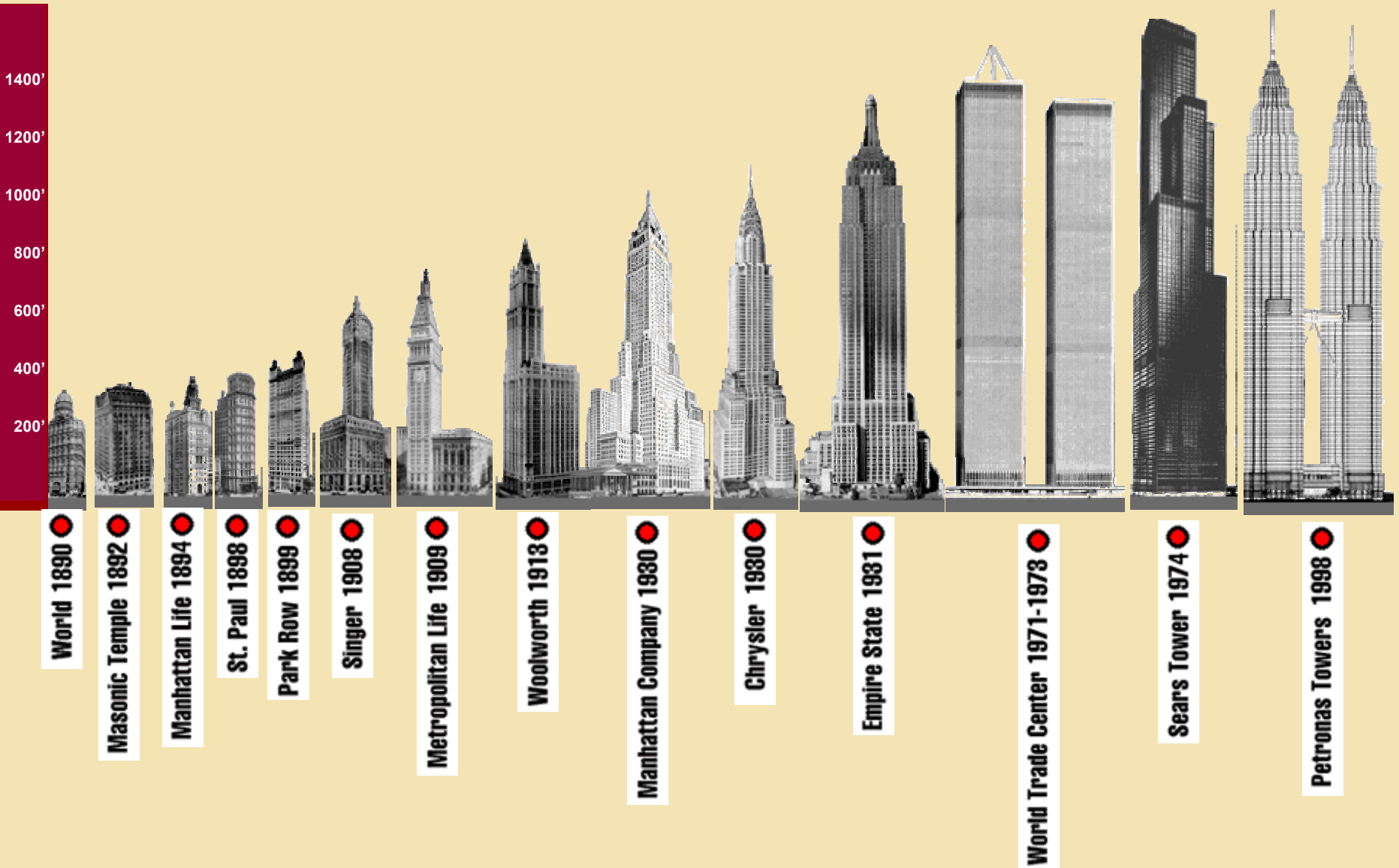
Opened: 1970-73; April 4, 1973 ribbon cutting



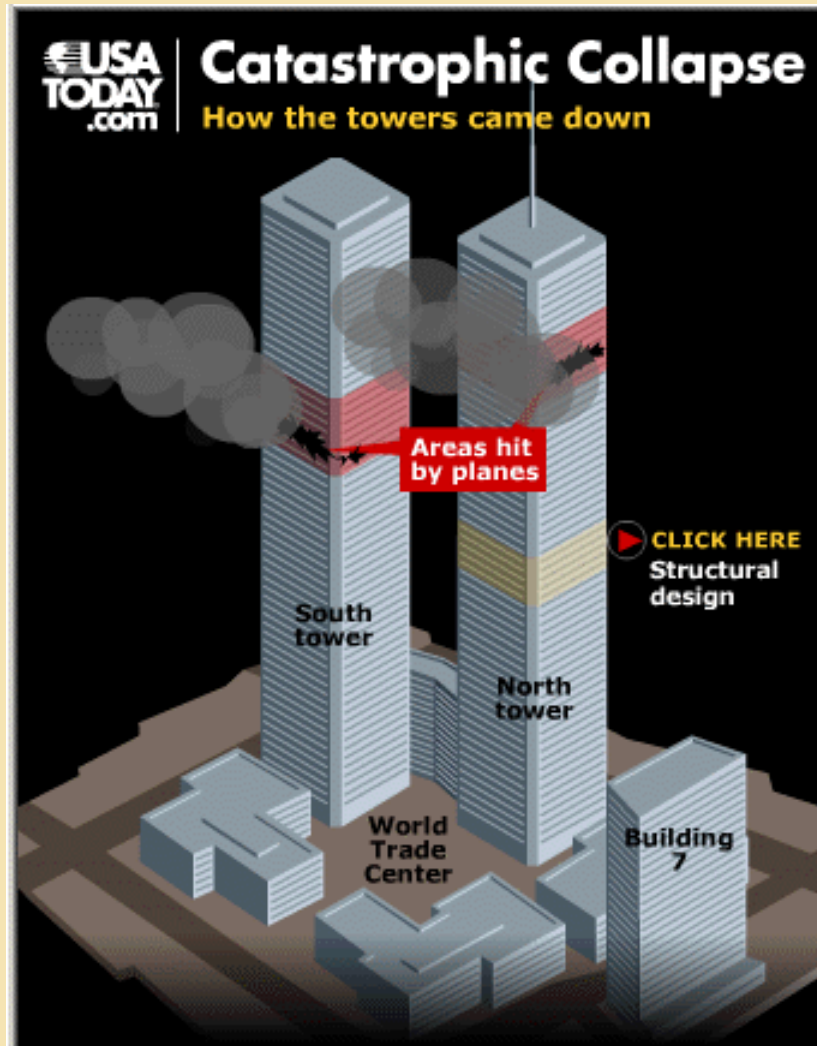
WTC Interesting Facts

- Construction cost an estimated \$1.5 billion.
- Engineers employed an innovative structural model: a rigid "hollow tube" of closely spaced steel columns with floor trusses extending across to a central core.
 - The columns were finished with an aluminum alloy to give a the silver-like coloring
- The twin towers were the first skyscraper buildings designed without any masonry.
- For the elevators to serve 110 stories with a traditional configuration would have required half the area of the lower stories be used for shaftways.
 - Elevators were designed such that passengers would change at "sky lobbies" on the 44th and 78th floors, halving the number of shaftways.

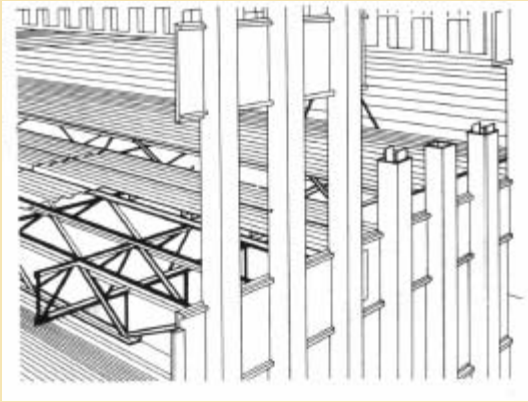
History of Skyscrapers



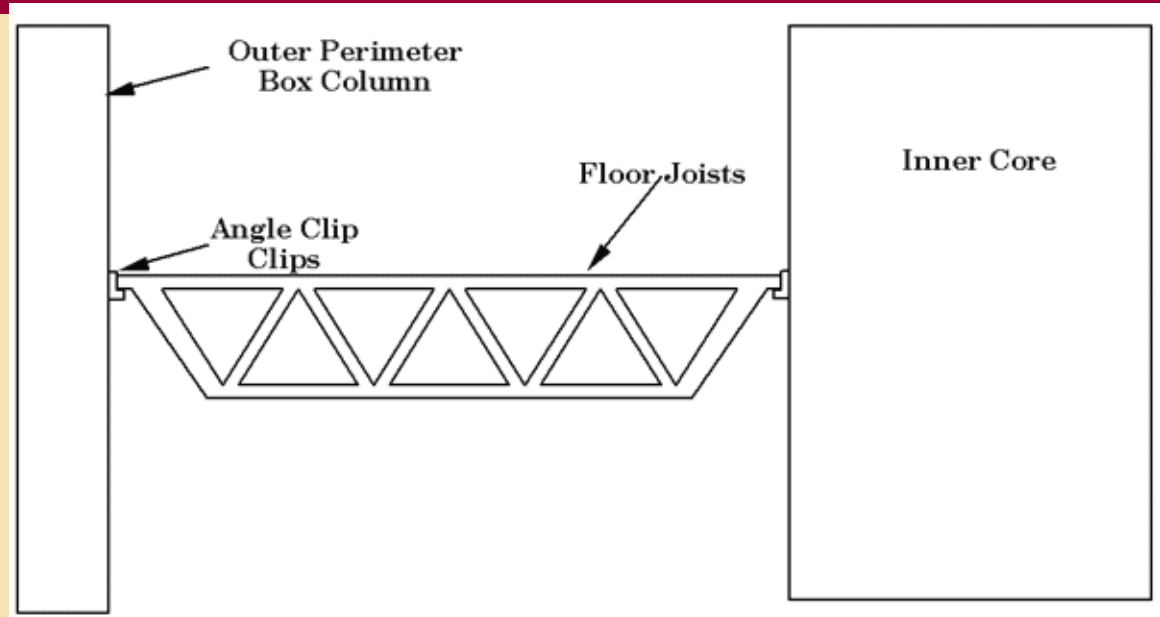
The Tragedy of September 11, 2001



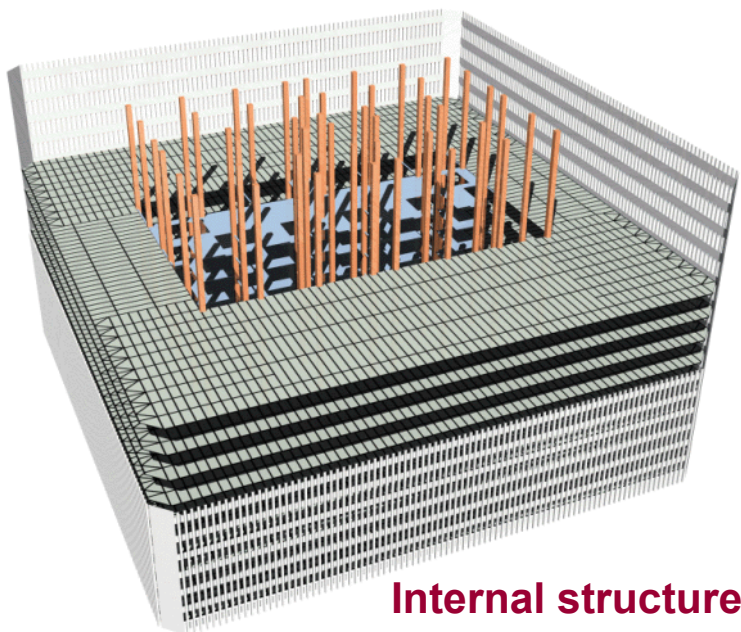
Schematic of Structure of WTC



External wall



Cross-section of WTC building structure



Internal structure

Boeing 767



Passengers – 375

Fuel Capacity – 23,980 gallons

Engines – PW 4062 63,300 lb thrust

GE CF6- 80C2B8F 63,500 lb

Cruise Speed at 35,000ft – 530 mph

Take-off Weight – 450,000 lbs

Energy of Impact vs. Fuel of Aircraft

Energy of Impact

$$\text{Kinetic energy} = \frac{1}{2}mv^2$$

$$\text{Mass} = 204 \times 10^3 \text{ kg}$$

$$v = 197 \text{ m/s}$$

$$\text{KE} = 39.6 \times 10^8 \text{ J}$$

Energy Associated with Fuel

Energy per
gallon of fuel $\sim 132 \times 10^6 \text{ J/gal}$

Energy of fuel

$$\begin{aligned} \text{in aircraft} &= [20,000 \text{ gal} \cdot 132 \times 10^6 \text{ J/gal}] \\ &= 2.64 \times 10^{12} \text{ J} \end{aligned}$$

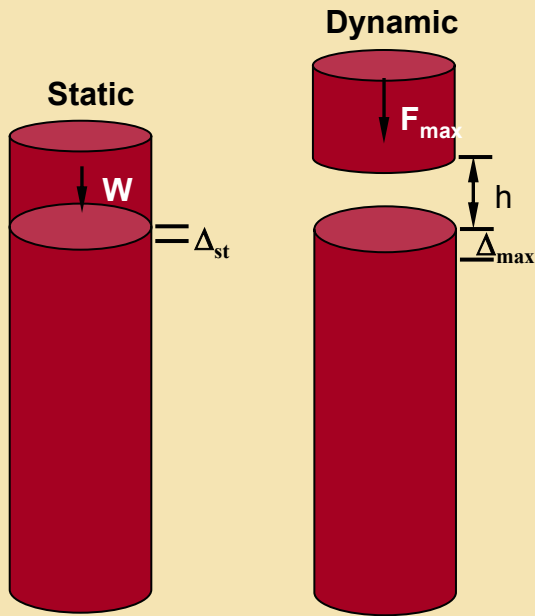
3 sticks of dynamite is 1 MJ, so

**energy content of the fuel $\sim 7,920,000$ sticks
of dynamite!**

Energy of impact was much lower than that of the burning fuel of the aircraft.

Engineering Analysis of WTC Collapse

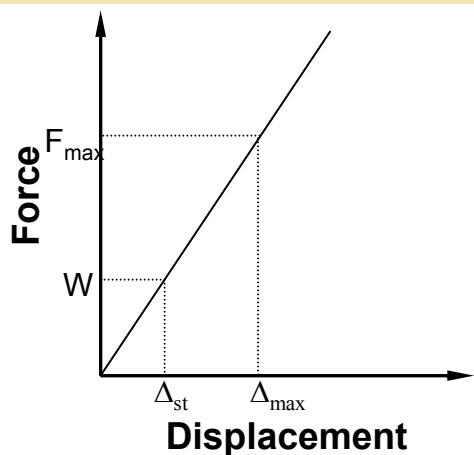
At the instant that the moving object strikes the stationary object:



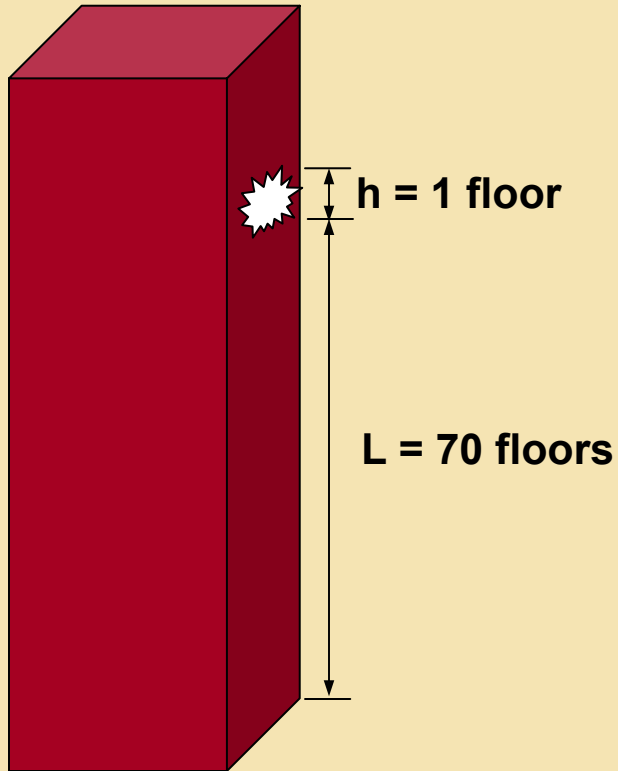
$$U_{\text{weight}} = U_{\text{strain}}$$

$$W(h + \Delta_{\text{max}}) = \frac{1}{2} F_{\text{max}} \Delta_{\text{max}}$$

$$F_{\text{max}} = 2W \left(1 + \frac{h}{\Delta_{\text{max}}} \right)$$



Engineering Analysis of WTC Collapse



$$F_{\max} = 2W \left(1 + \frac{1}{\epsilon_{\max}} \cdot \frac{h}{L} \right)$$

Assume creeping and softened steel yields at
 $\epsilon_{\max} = 0.001$

$$= 2Mg \left(1 + \frac{1}{0.001} \cdot \frac{1}{70} \right)$$

$$= 30.6 Mg$$

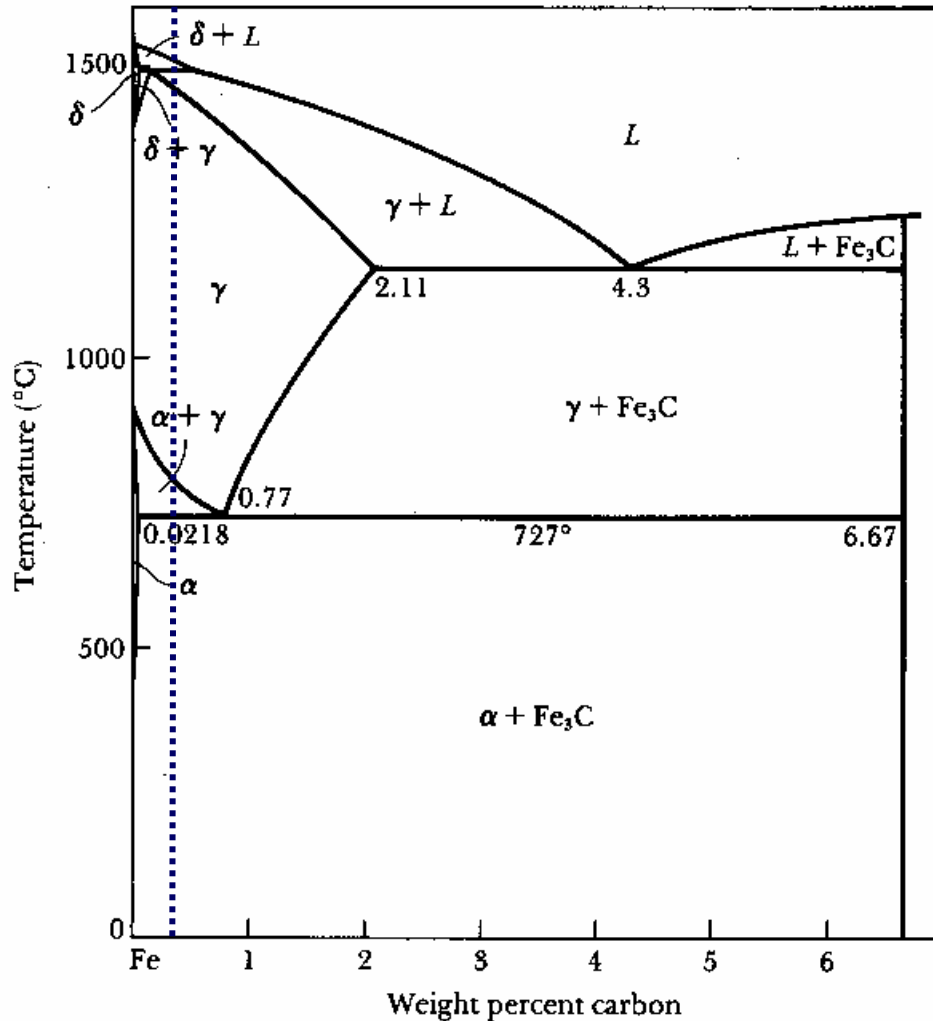
Impact force is at least 30 times mass of floors above impact!

Composition and Properties of A36 Steel Used in Core of WTC

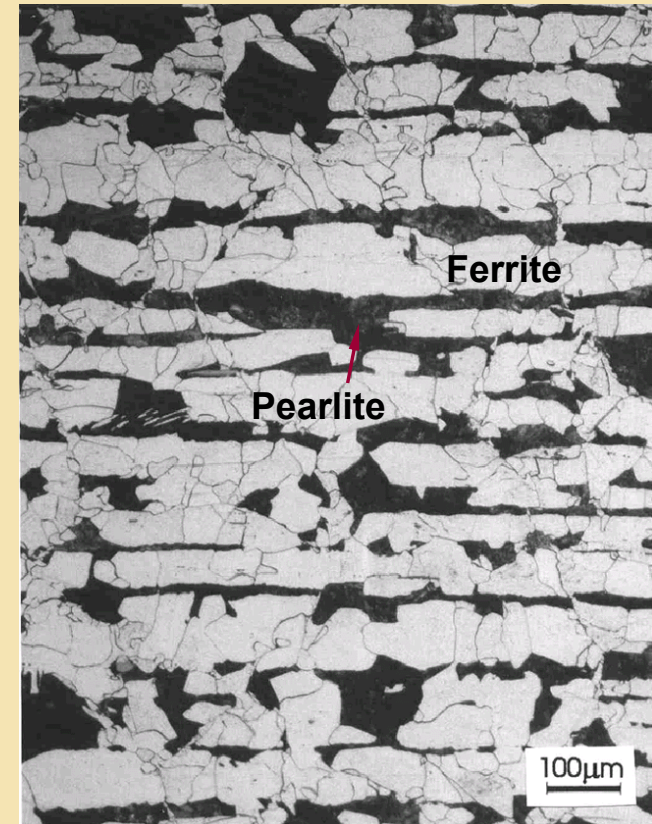
0.30 %	C
0.15-0.3%	Si
0.8-1.2%	Mn
0.04%	P
0.05%	S
Balance	Fe

Tensile Strength	Yield Strength	Elongation (50 mm gage length)
400-550 MPa	220-250 MPa	23%

Fe-Fe₃C Phase Diagram

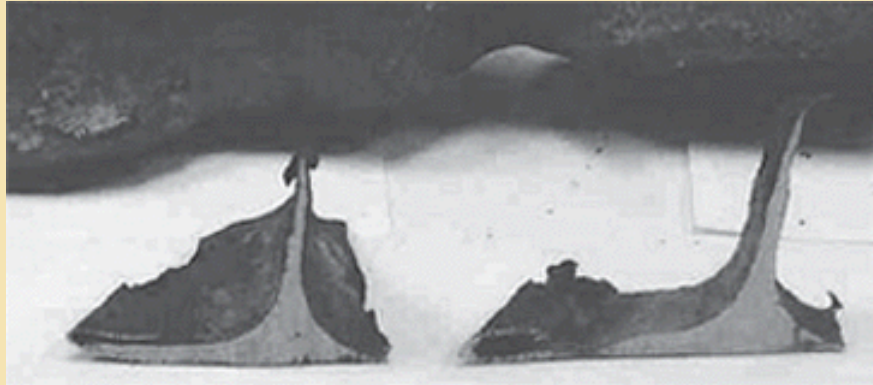


Microstructure of Unaffected A36 Steel

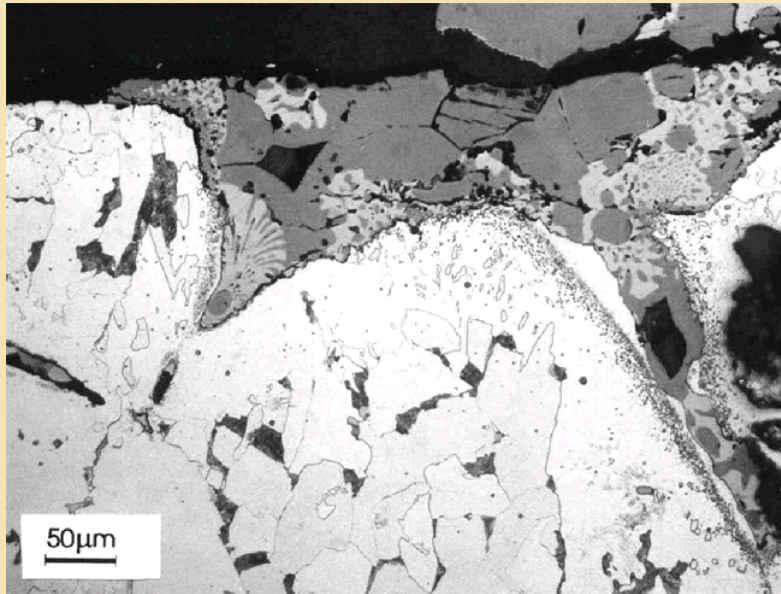


Banding of ferrite and pearlite are observed due to hot working of plate

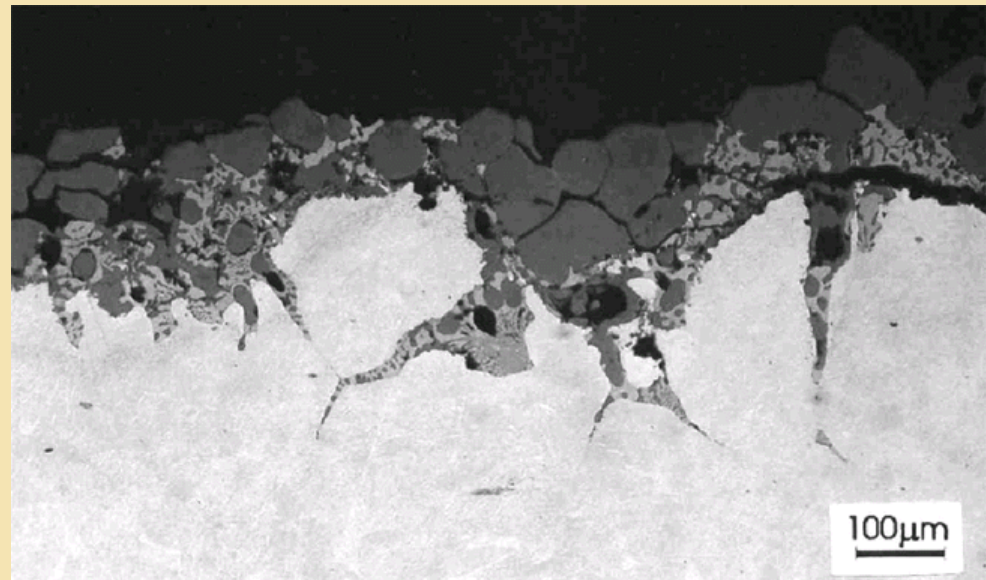
I-Beam Cross-Sections of A36 Steel from WTC 7



The beams appear severely eroded.

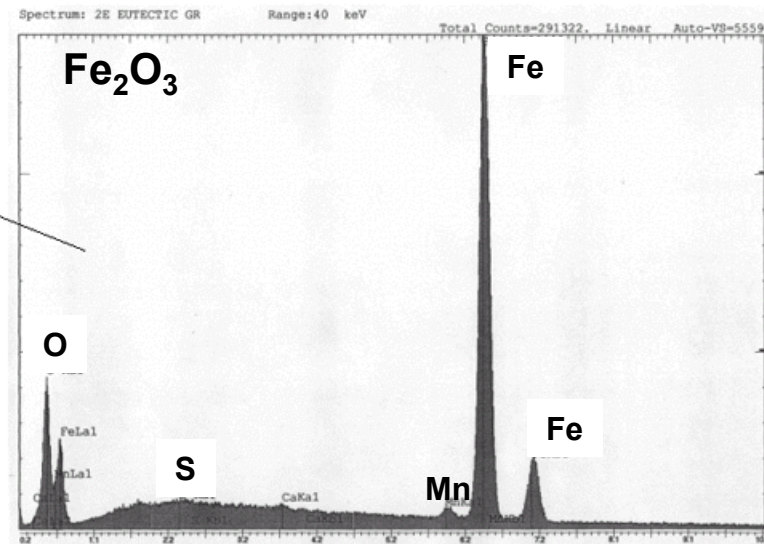
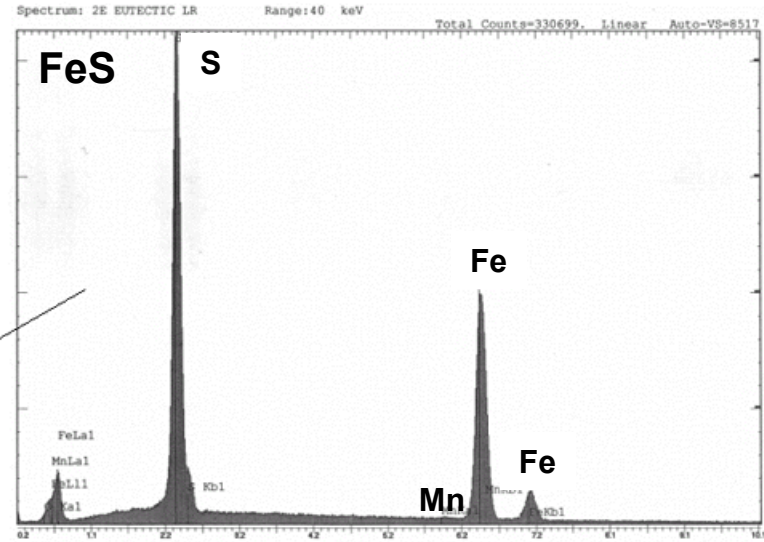
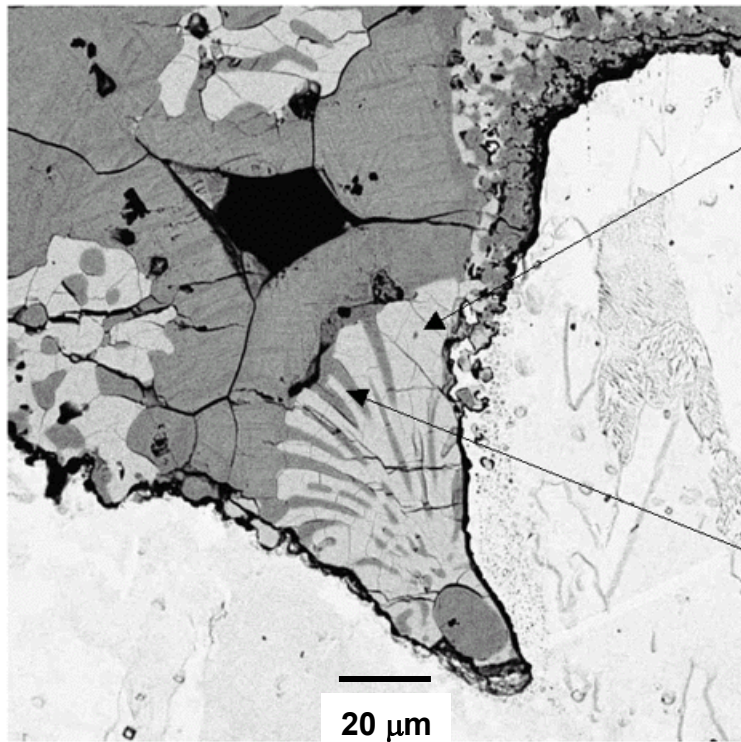


Surface microstructure appears to be eutectic mixture of FeS and Fe₂O₃



Severe oxidation and intergranular melting is observed

Composition Analysis of A36 Steel from WTC-7

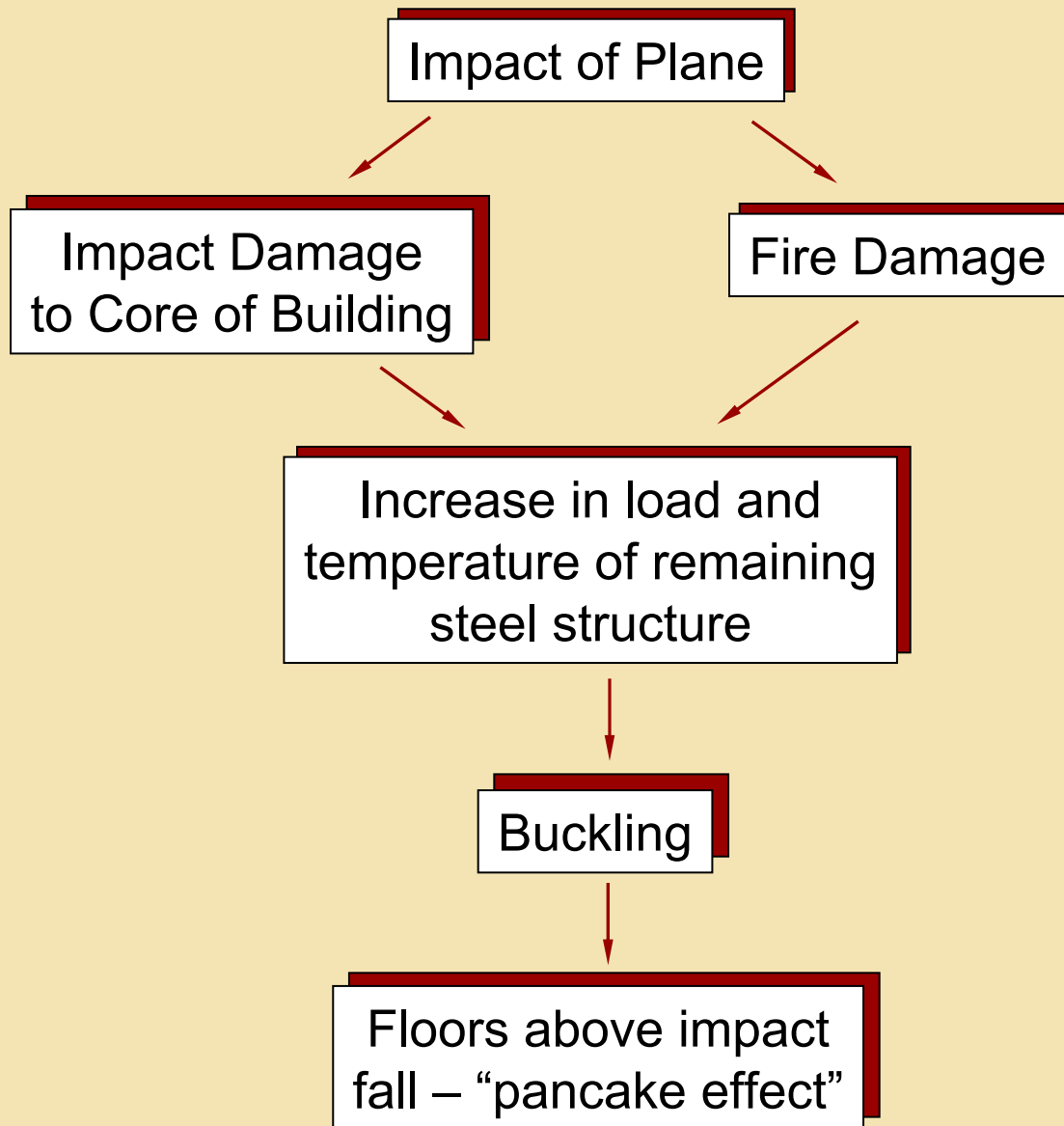


Preliminary Mechanical Testing of WTC Steel

- National Institute of Standards and Technology have conducted preliminary tests on 236 pieces of steel from WTC wreckage.
- Requirement for tensile strength of steel was ~ 36,000 psi.
 - NIST tests showed steel to be capable of bearing ~ 42,000 psi.

Steel beams from the World Trade Center generally met or exceeded design strength requirements.

Proposed Sequence of Events that Caused Collapse of WTC Buildings



Thoughts and Speculation about Failure

- Buildings were meant to withstand impact of a Boeing 707 – same amount of fuels as 767
- WTC fire was fuel rich (not typical of office fires) – smoke was dark black
 - Temperatures of fuel rich fires are typically $< 827^{\circ}\text{C}$)
- Steel did not melt, but may have been in the austenitic phase field (above eutectoid temperature of 727°C).
 - Severe weakening due to creep.
- Thermal stresses may also have played a role.
 - Steel was cool from outside and quite hot inside
- Floors above impact may have caused significant damage to steel joints, with very falling floor, during collapse.

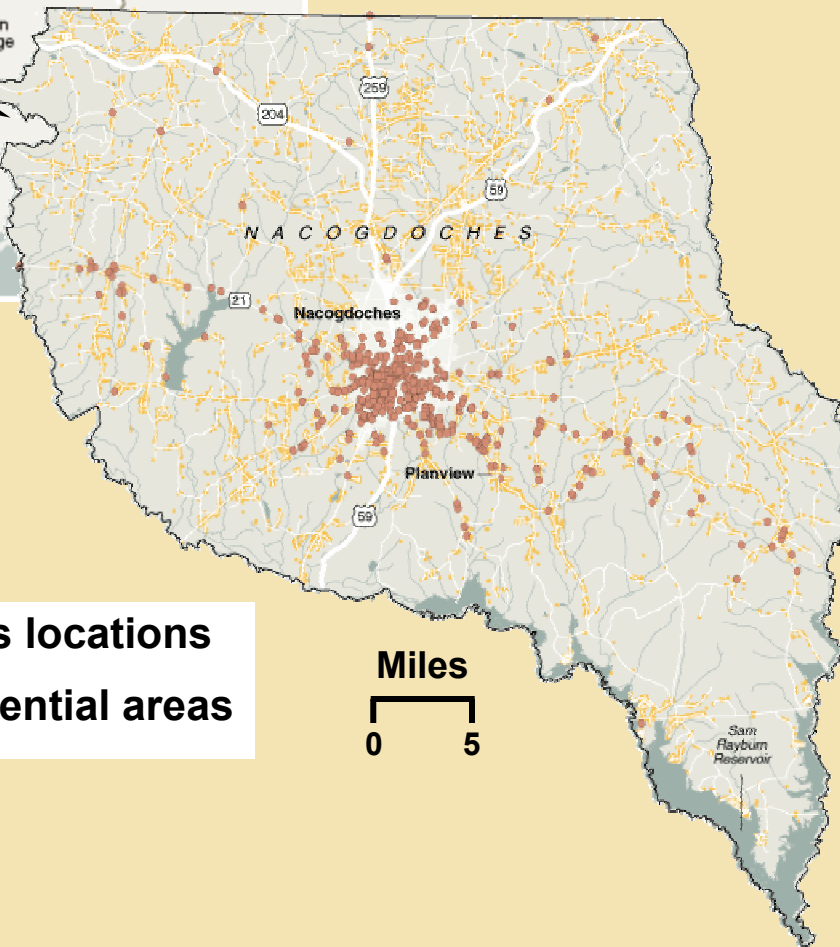
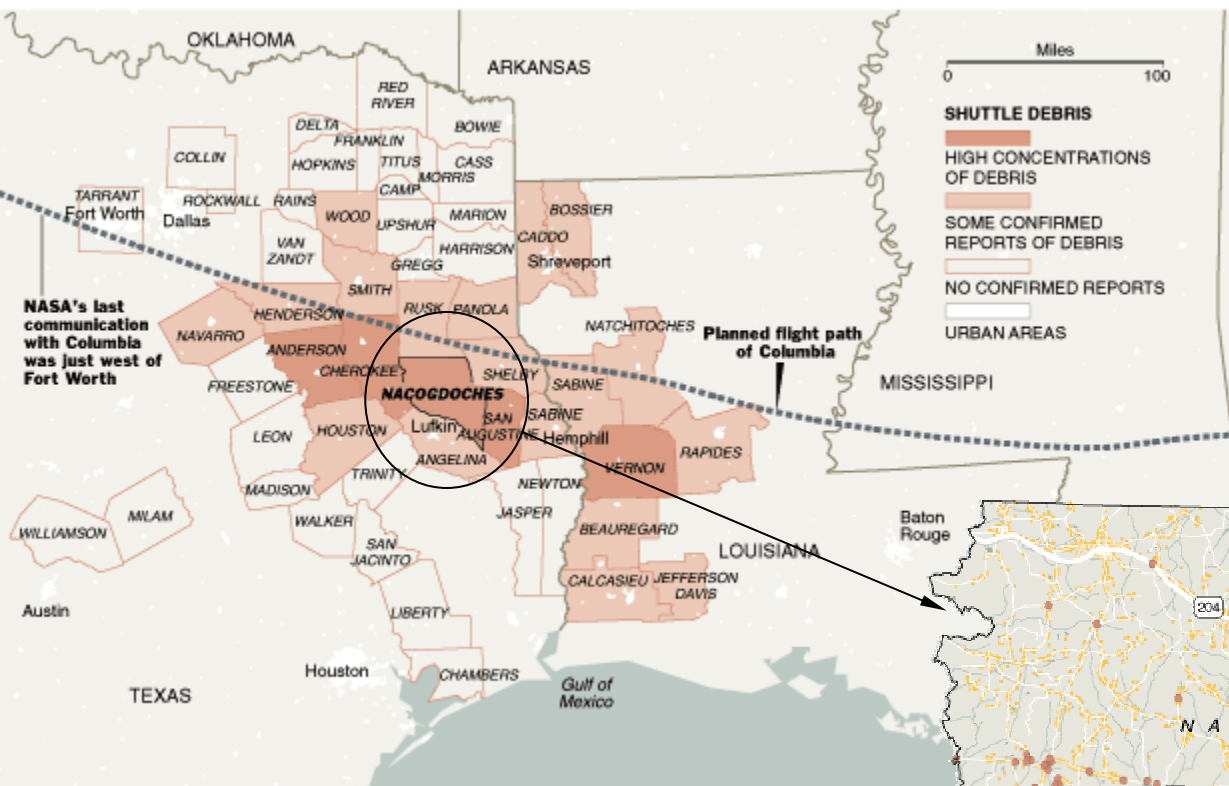
Collapse of the WTC building would not likely have been prevented by better design.

The Columbia Space Shuttle Disaster



Planned Re-entry Path of Shuttle

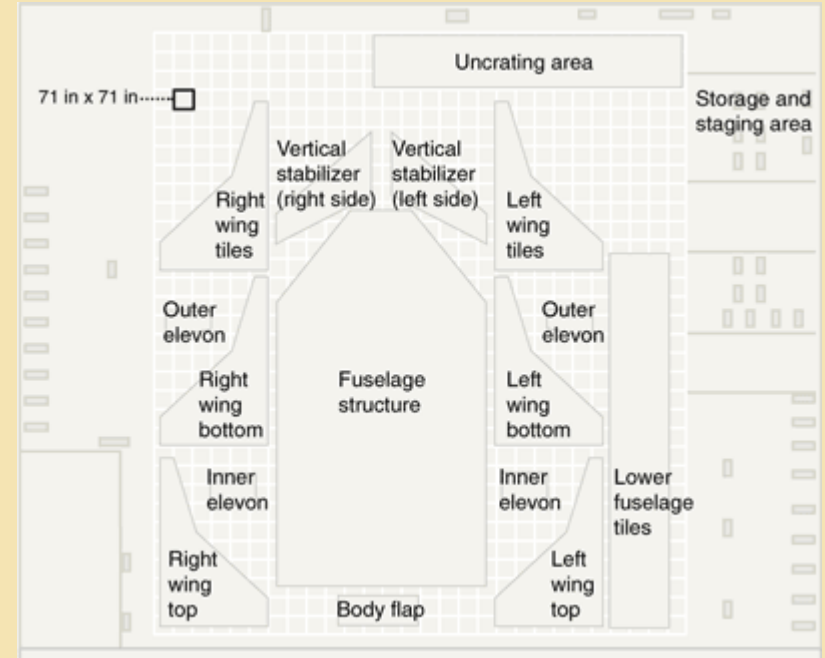
Map of Debris Field



Reassembling the Shuttle

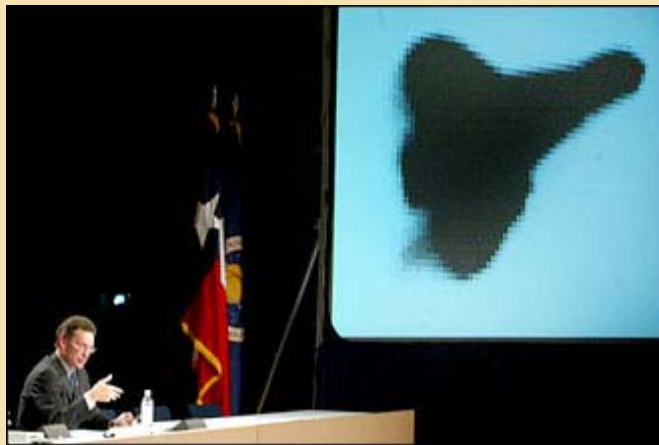


Reassembly of Shuttle at NASA hangar



Schematic of shuttle parts for reassembly

Satellite Photograph of Shuttle in Space

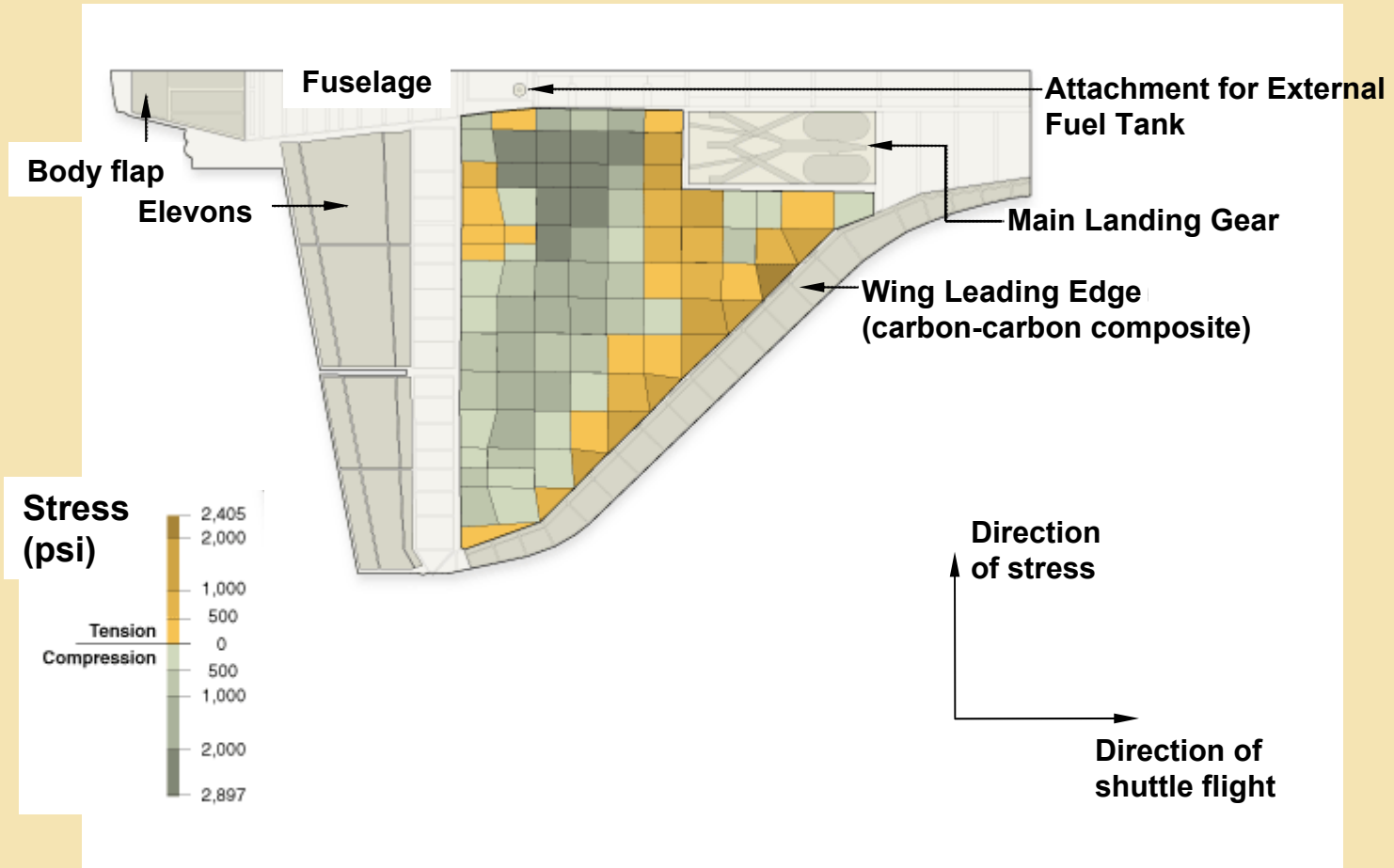


**NASA Briefing by
Ron Dittmore, Shuttle Program Manager**

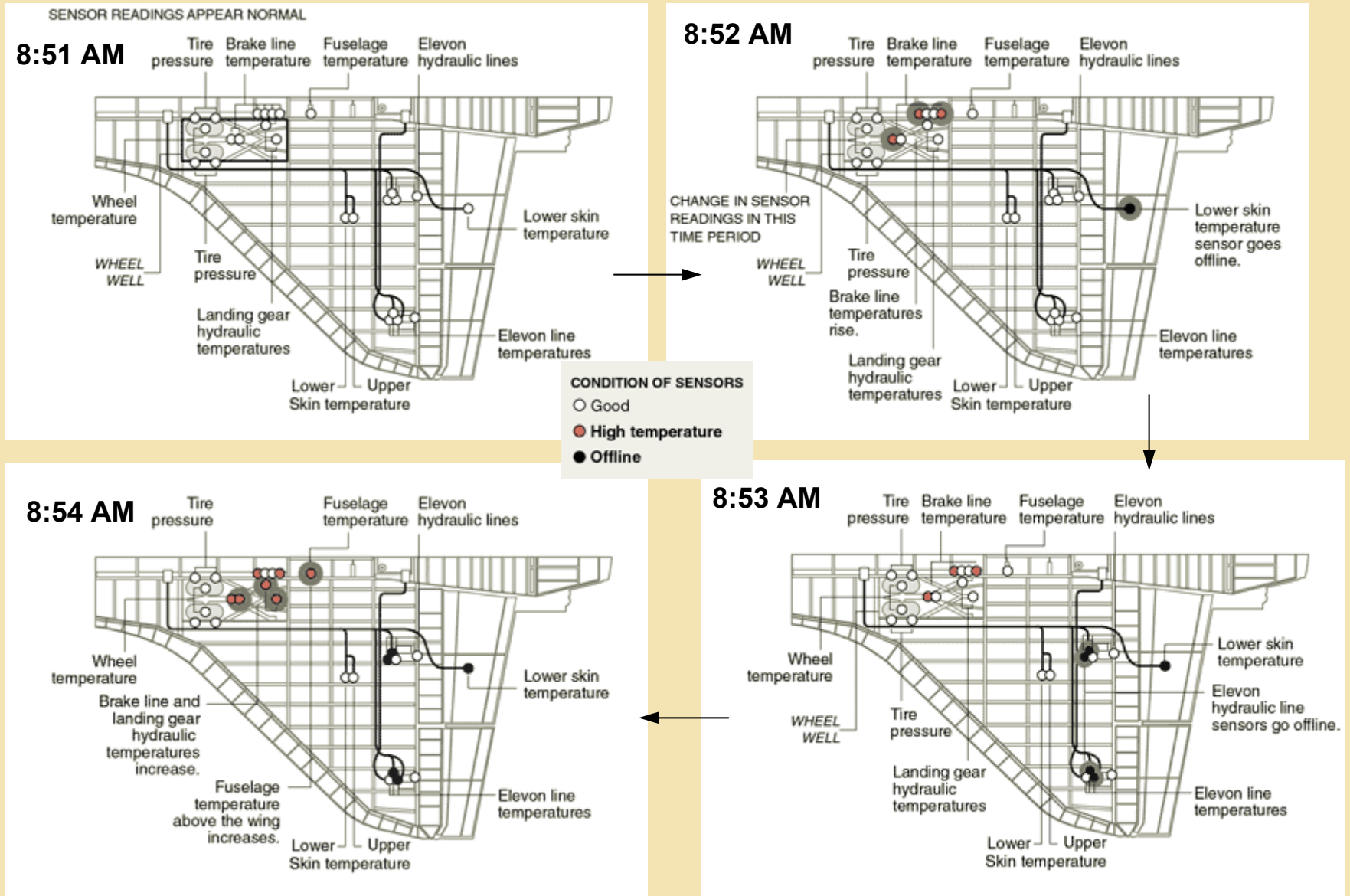


**What is the
protrusion on the
leading edge of
the left wing?**

Thermal Stresses in Shuttle Wing During Re-entry

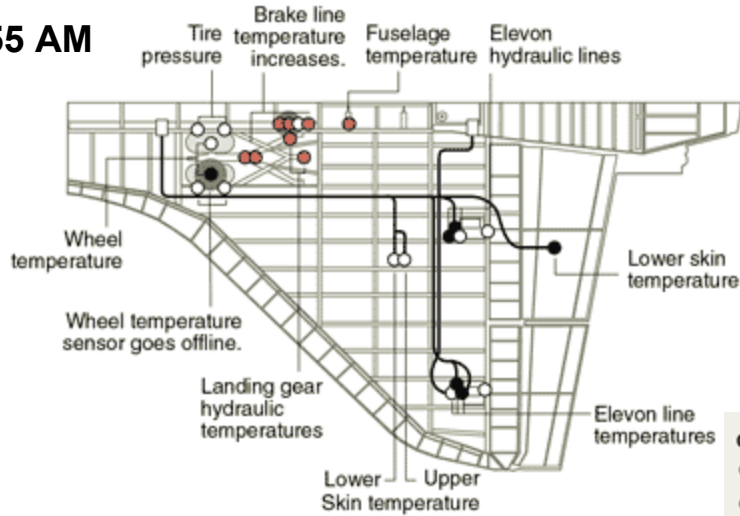


Temperature Evolution in Left Wing

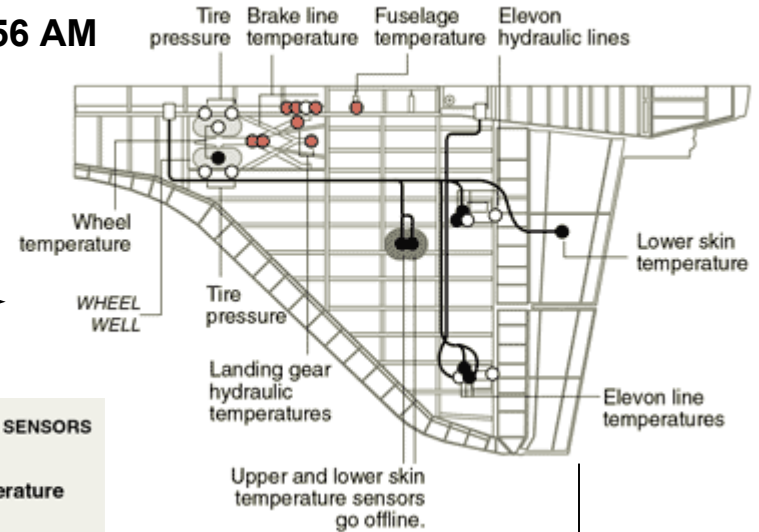


Temperature Evolution in Left Wing

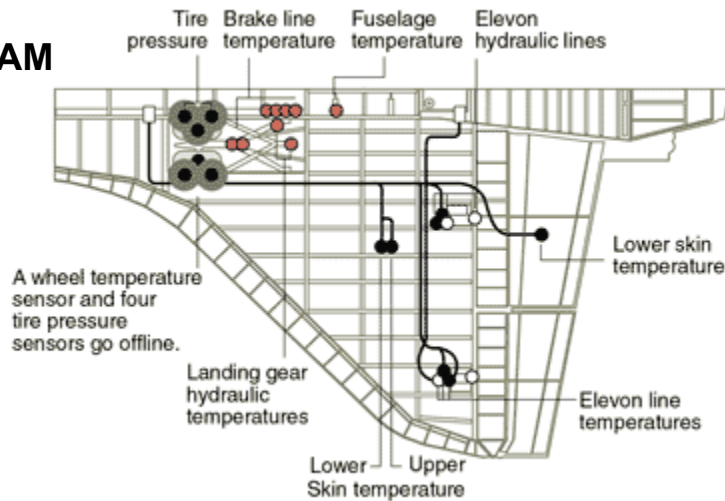
8:55 AM



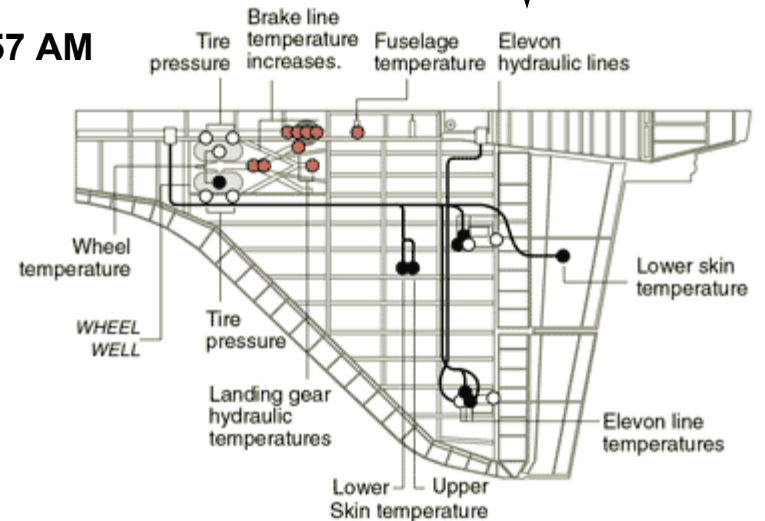
8:56 AM



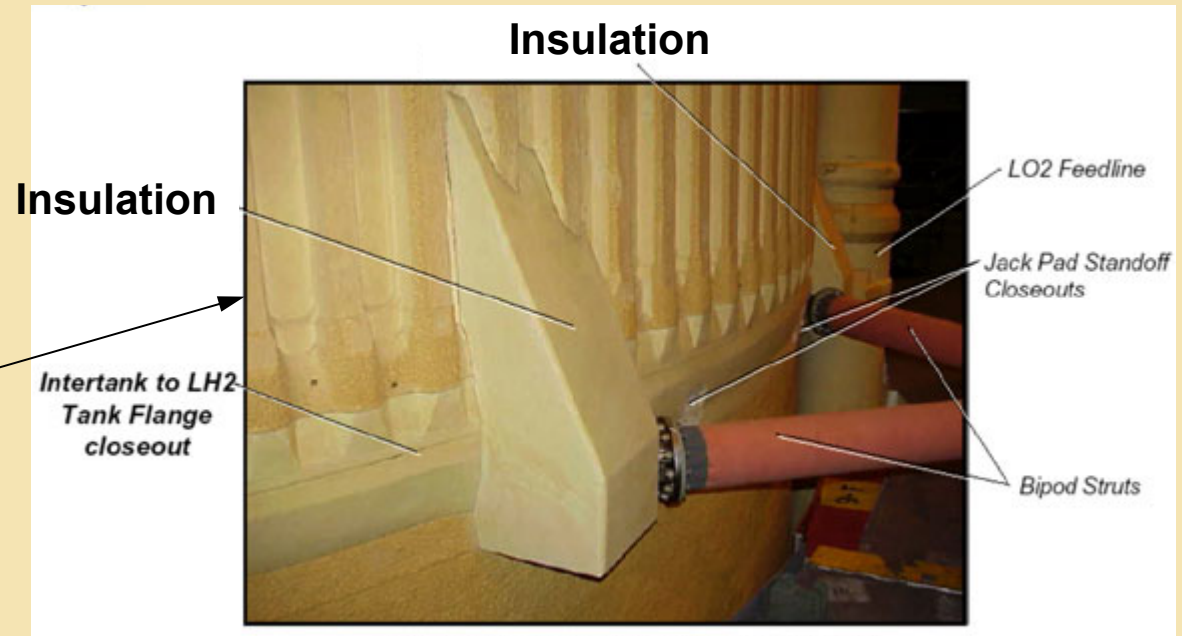
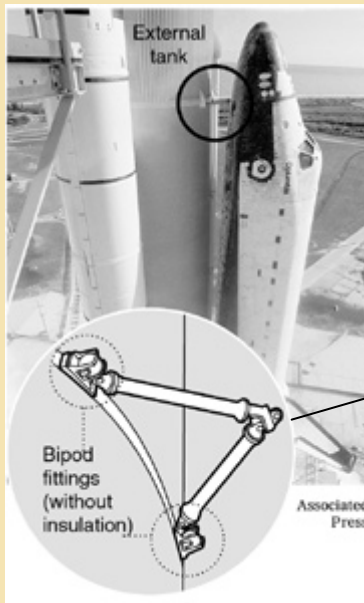
8:58 AM



8:57 AM



Schematic and Photograph of Bipod/Flange Area with Insulation



Bipod/Flange Area

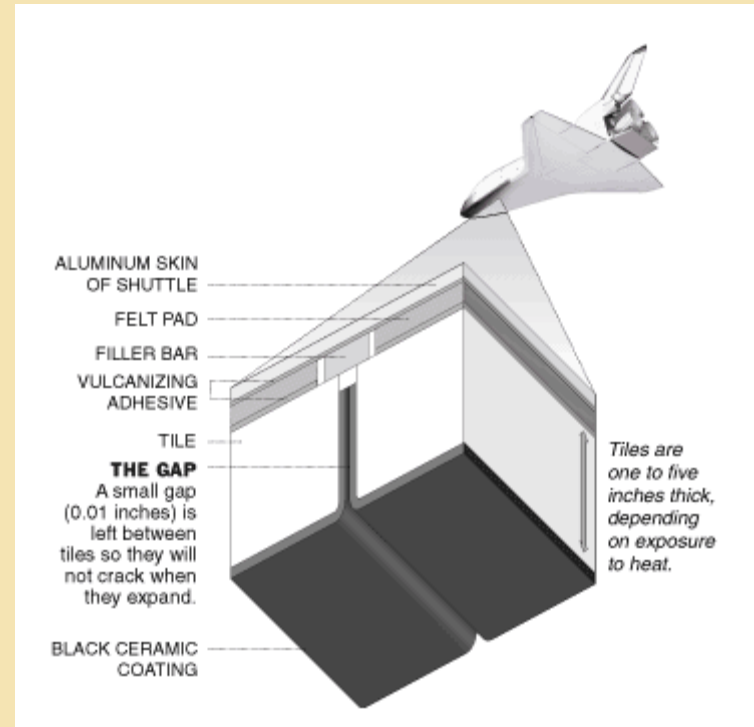
Foam Striking Left Wing of Shuttle



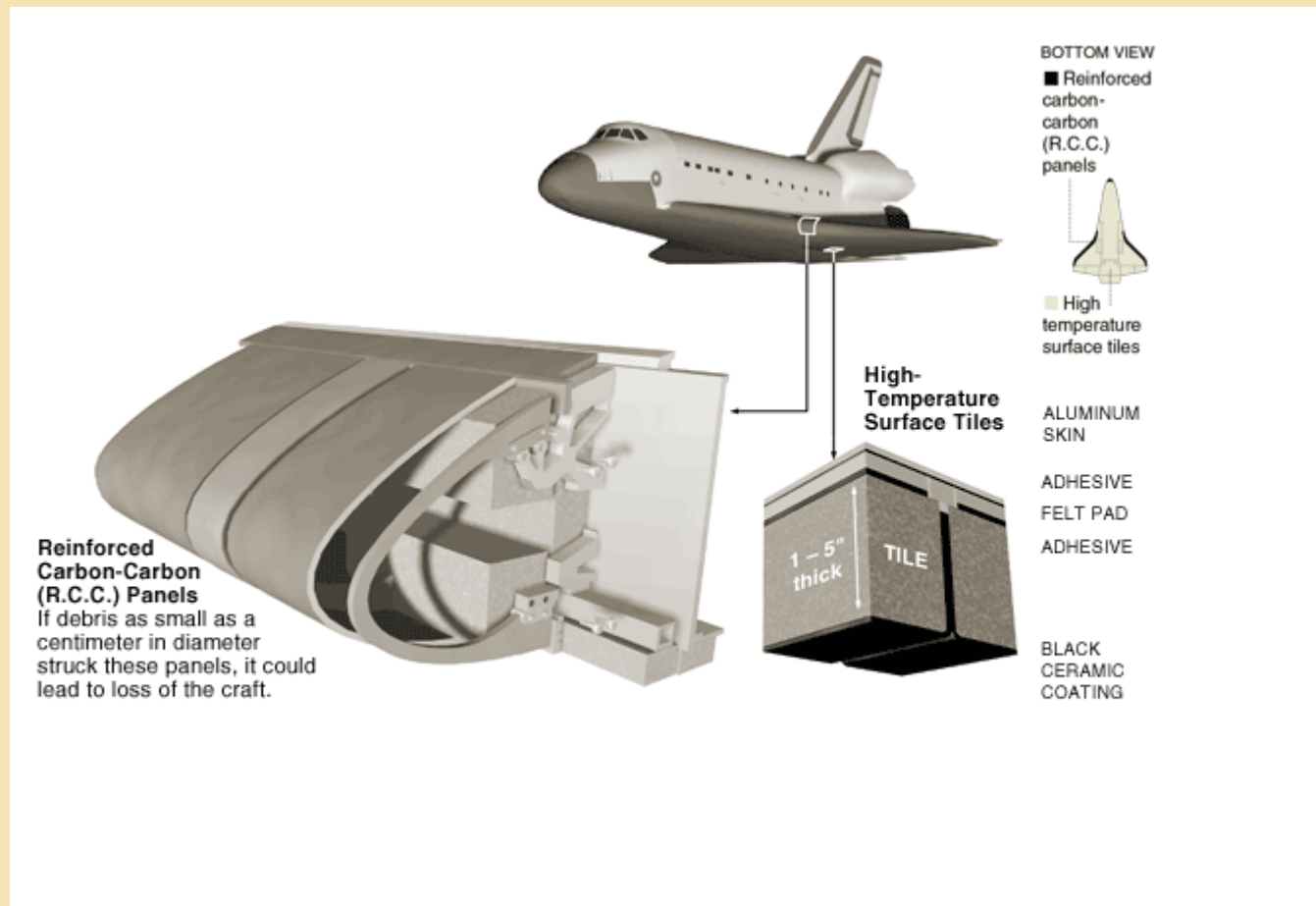
NASA Flight Video



Ceramic Tile used for Thermal Insulation



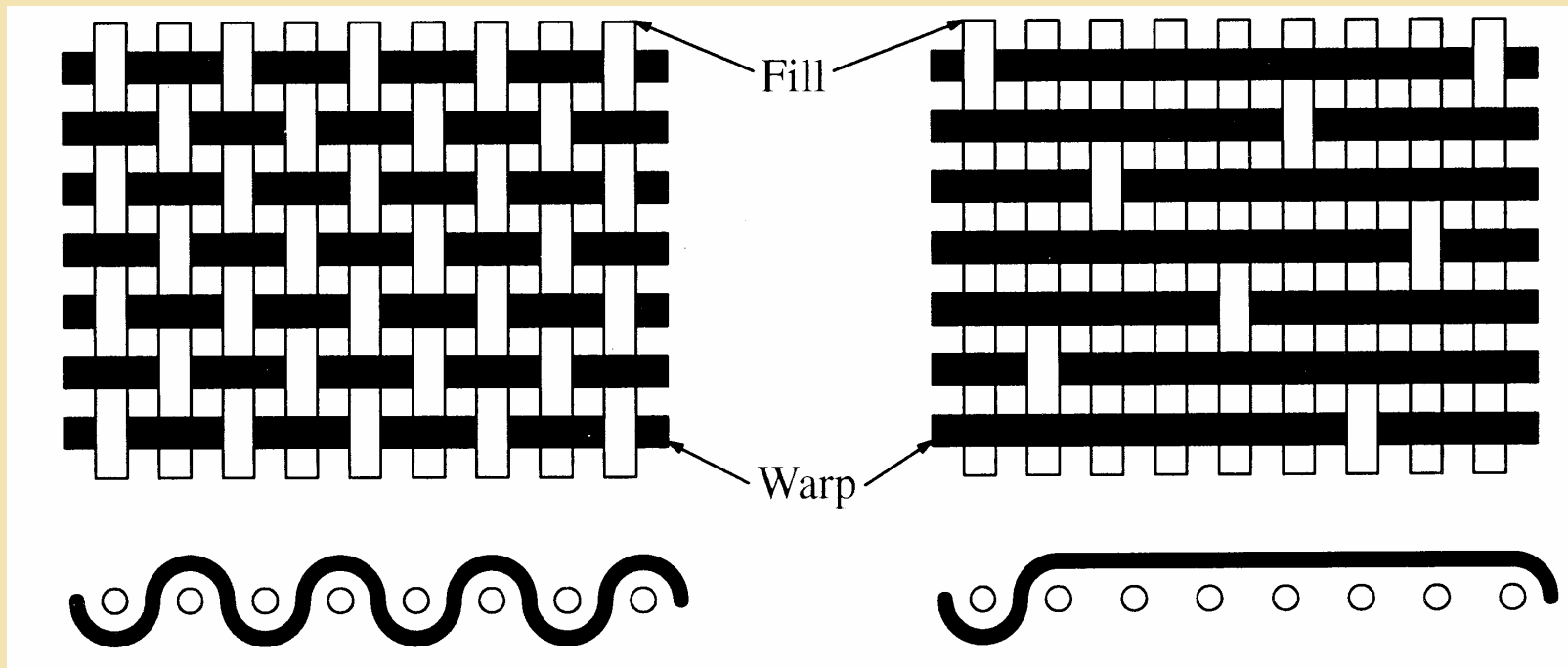
Carbon/Carbon Composite in Leading Edge of Wing



Woven Fiber Fabrics

Plain

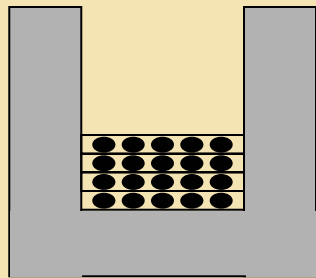
Eight Harness Satin



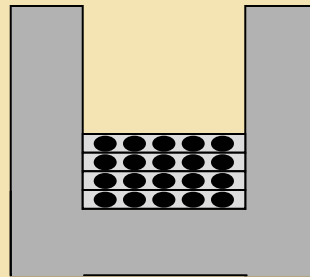
- stable
- handleable
- easier processing

- drapeable(contours)
- distinct faces
- less porous

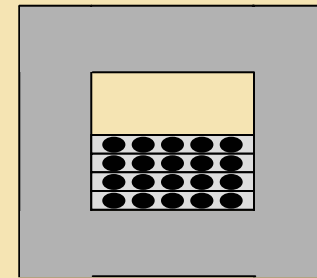
Processing of Carbon/Carbon Composites



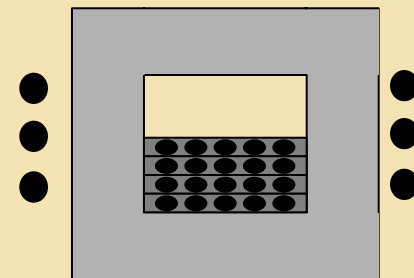
Fiber laminate stacking



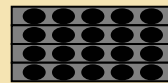
Resin or pitch impregnation



Evacuate and seal



**Hot isostatic pressing
+ carbonization**



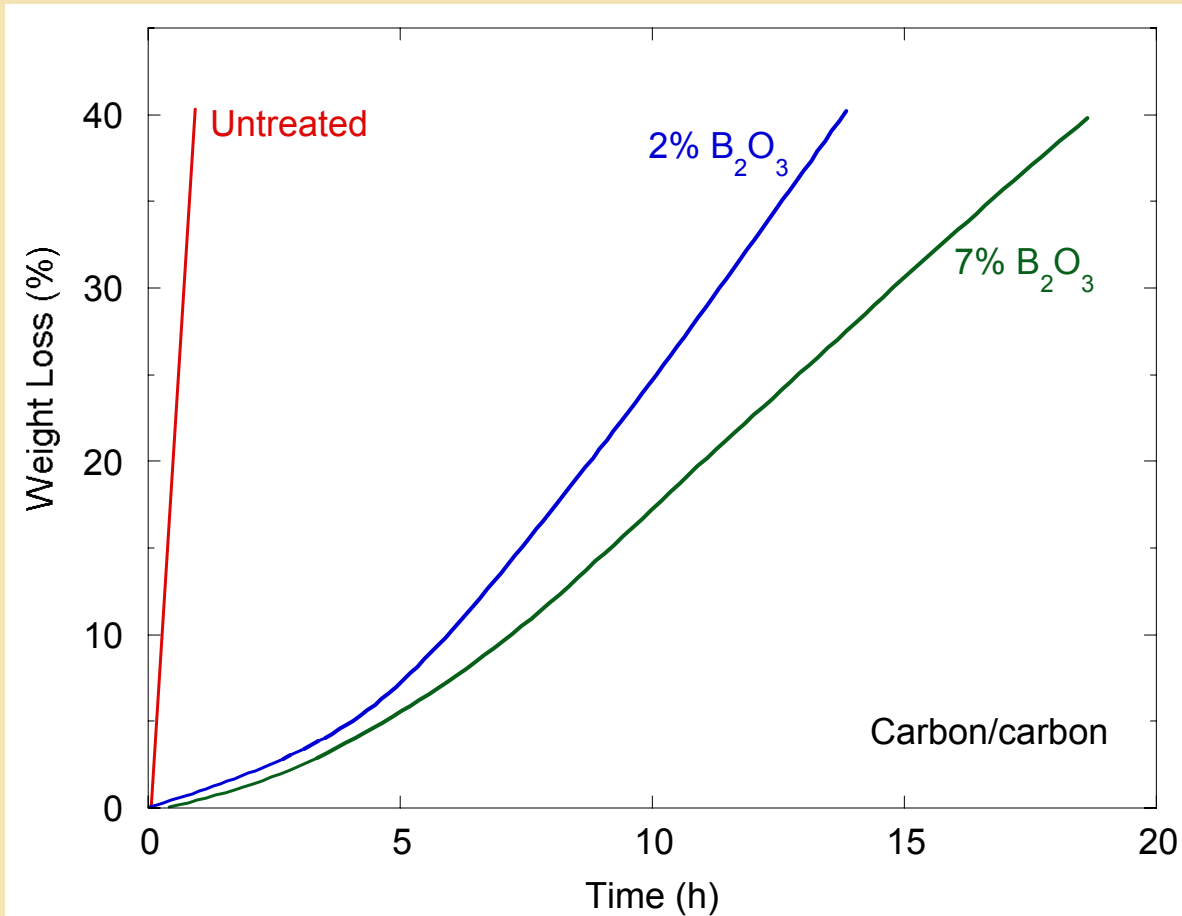
**Carbon/Carbon
composite**

Thermal Protection Design for Space Shuttle

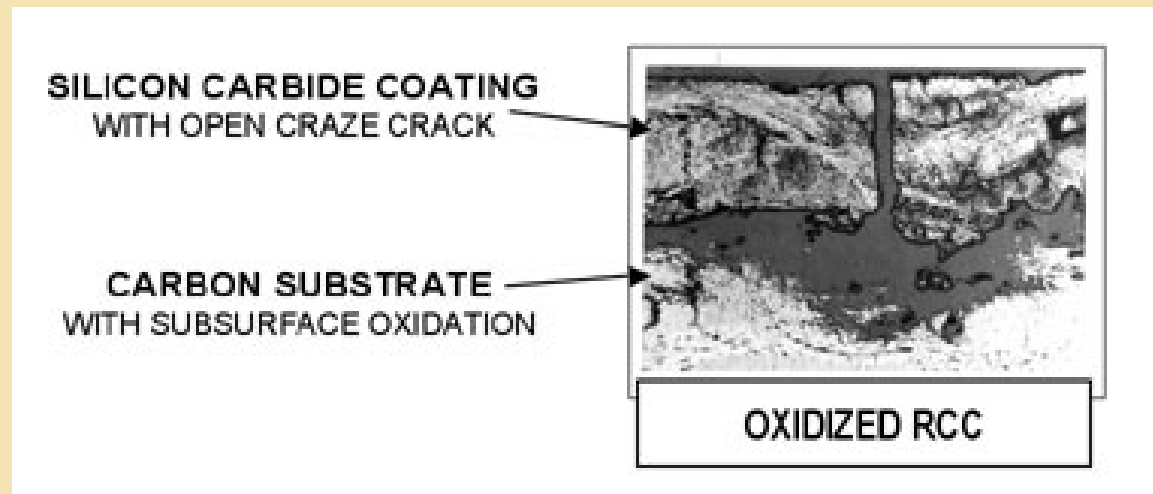
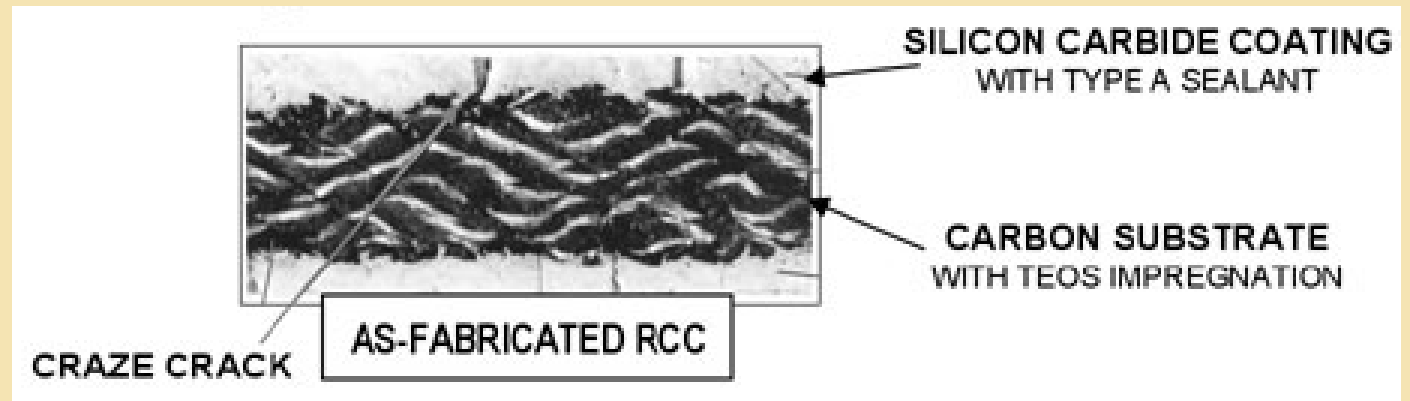


Borosilicate Overglaze

Glass formers penetrate and seal any cracks in outer coating

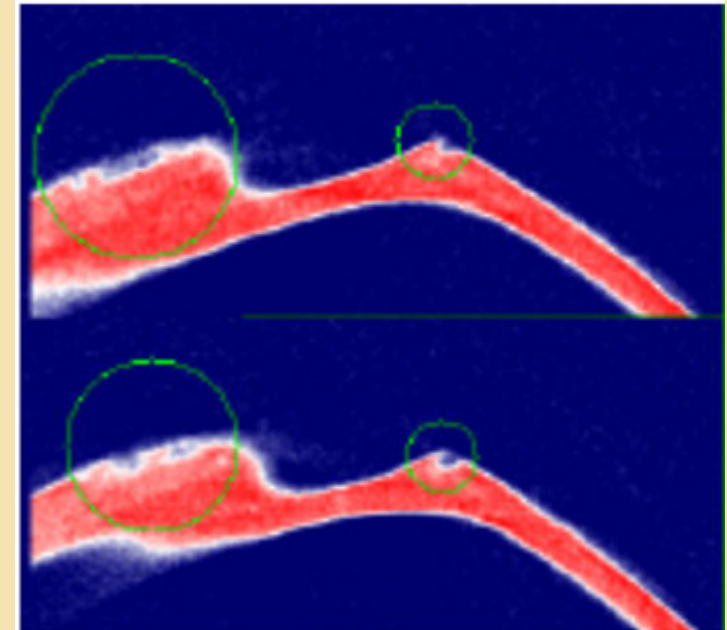


Carbon-Carbon Composite Used in Leading Edge of Shuttle Wing



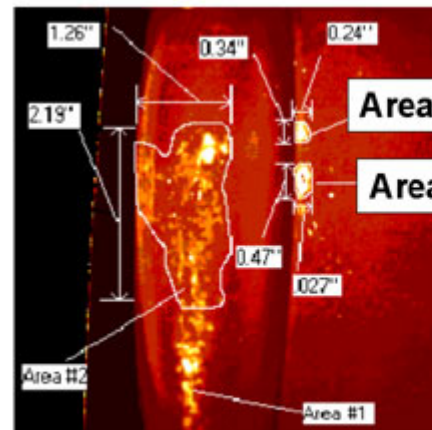
Foreign Object Damage in Previous Shuttle Mission

- Damage was on the upper edge measuring 0.10"L x 0.15"W
- Defect had large cavity with carbon substrate oxidation

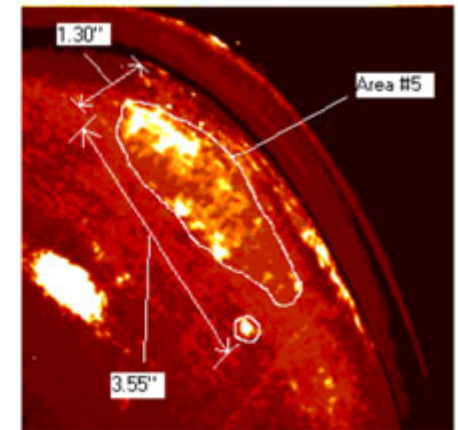


CT Scan (Tomography)

Space Shuttle Atlantis – 11/2002



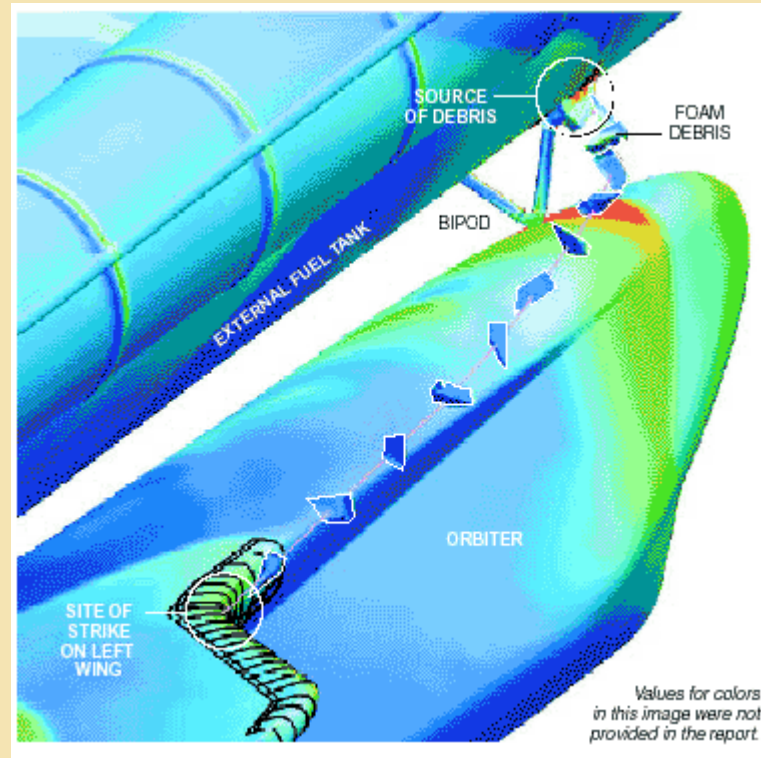
Front View



Side View

Oxidation flaws

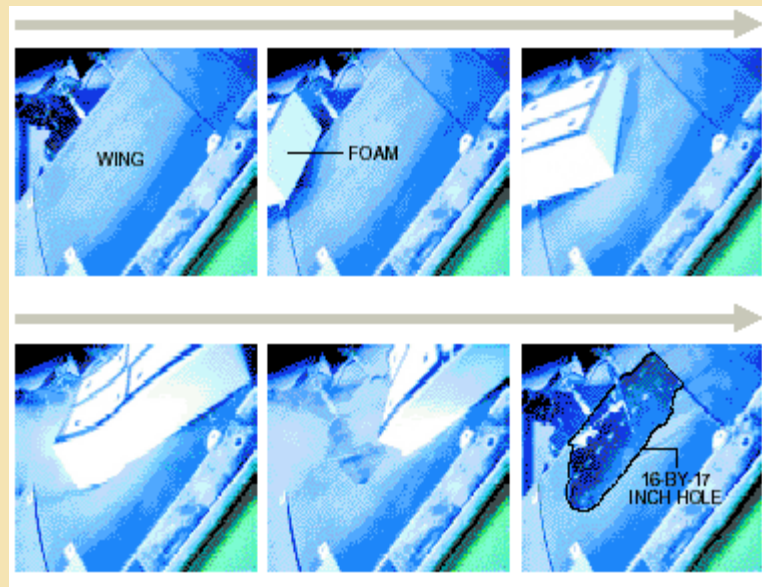
Summary of Damage Sequence



81.9 seconds after liftoff, debris hits wing

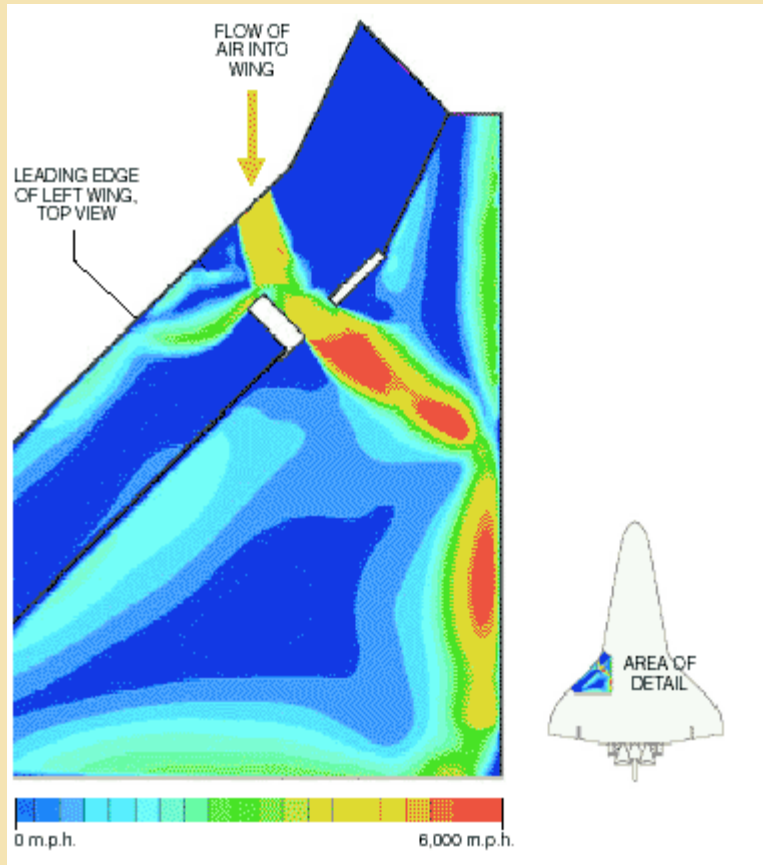
Computer models predicted velocity of foam (~1.7 lbs.) to be ~ 500 miles per hour.

Estimating the Damage During Impact

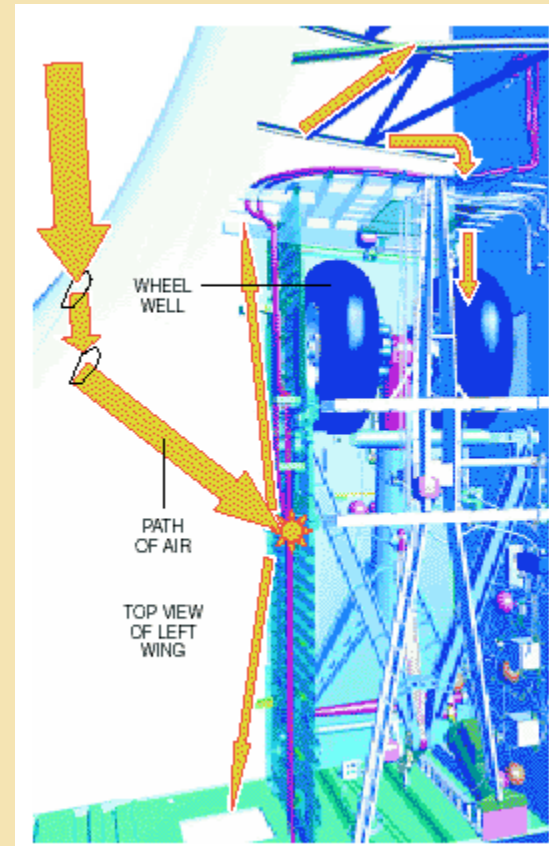


Foam impact testing on wing of shuttle Atlantis

Summary of Damage Sequence (contd.)



Speed of air moving through wing

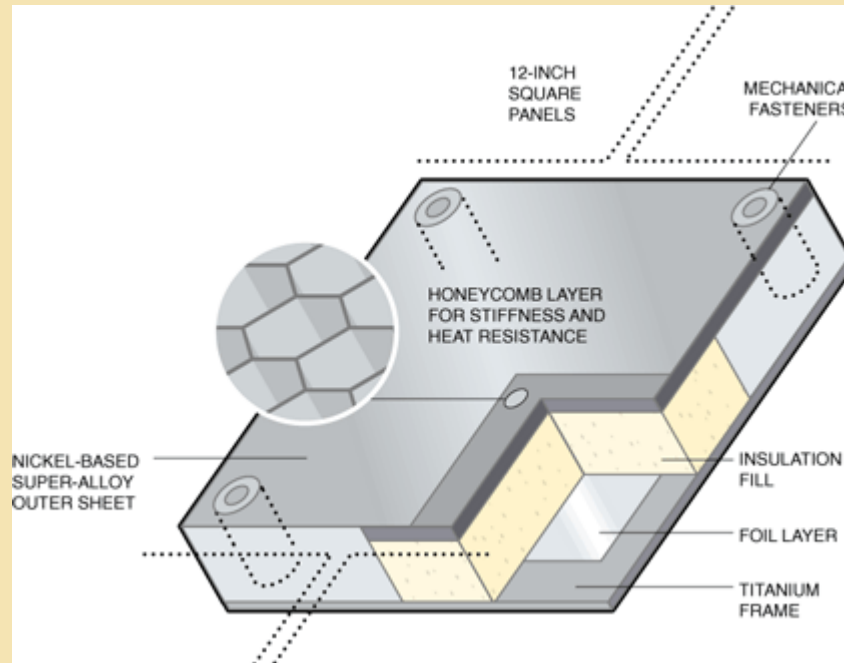


**Superheated air flowed through wheel well
- Al trusses melted, weakening shuttle structure**

Lessons Learned from Shuttle Columbia Explosion

- Space shuttle tiles may not be adequate for impact damage resistance
- Carbon/carbon composite leading edge should be tailored for higher impact damage/oxidation resistance
 - More frequent inspections (after flights) of leading edge should be conducted
- Impact of foam should be addressed in future flights
- Emergency escape system should be devised and incorporated into flight plans

Alternate Tile Material/Design for Space Shuttle



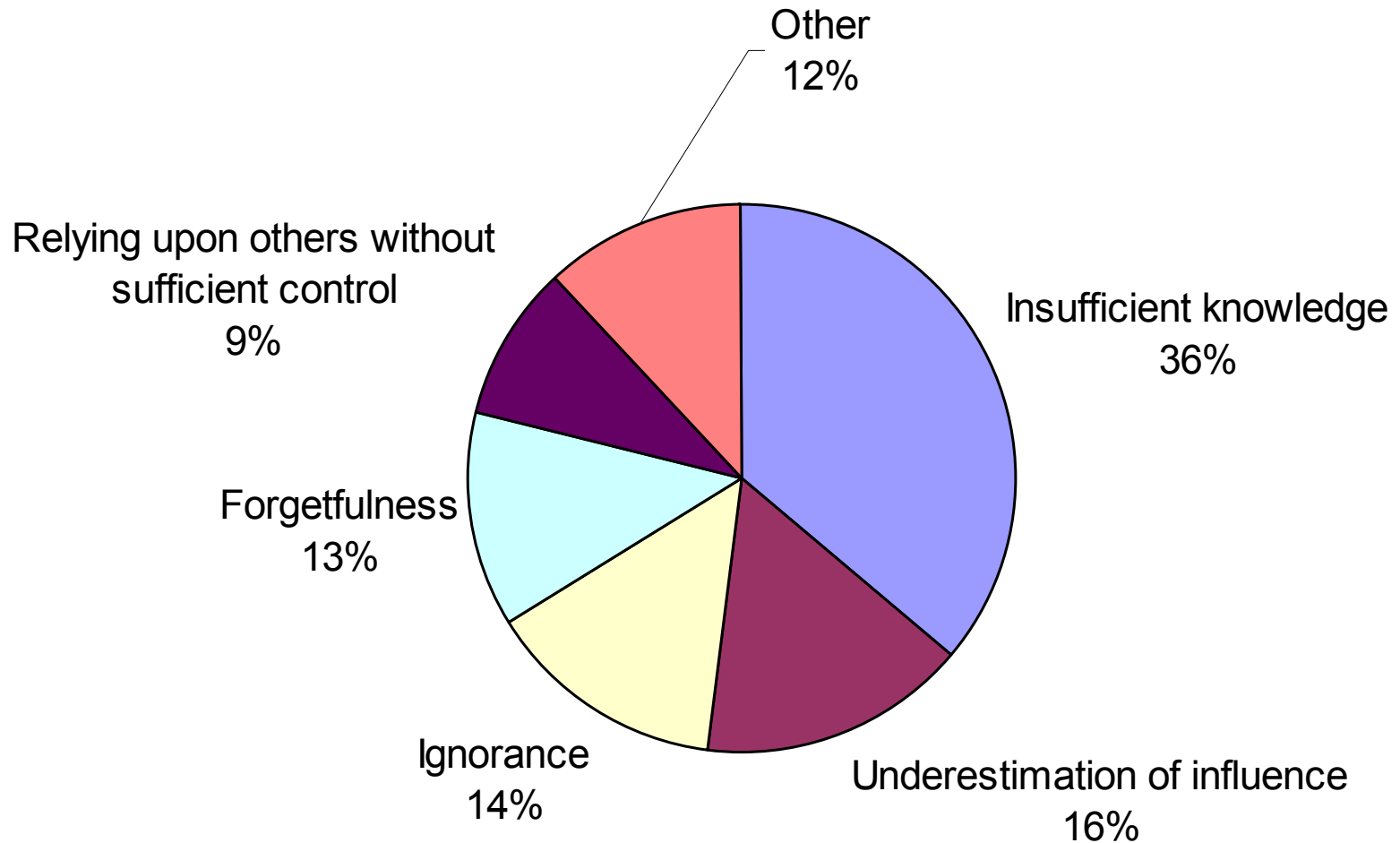
Shuttle Accident Board Recommendations for Resuming Flights

- Begin aggressive program to eliminate all shedding of debris from external fuel tank.
 - Increase shuttle's ability to sustain minor debris damage.
- Make it possible to inspect damage to shuttle's thermal protection system and make emergency repairs.
- Upgrade imaging system so it can provide at more useful views of shuttle at liftoff.
- Obtain high-resolution images of external tank after it separates, and images of wings' leading edge and thermal protection system.
- Develop a comprehensive inspection plan to determine the structural integrity of components made of reinforced carbon-carbon.

General Observations for Materials Engineers

- Continue to conduct research/engineering of structural materials.
 - Microstructure-property relationships are the key to understanding material behavior!
- Communicate and collaborate with engineers from other disciplines (civil, mechanical engineers).
- Make general public and public officials aware of the importance of engineering materials to our society (outreach).

To Engineer is Human



Study by the Swiss Federal Institute of Technology – Zurich

800 cases of structural failure, 504 killed, 592 injured, millions of dollars in damage incurred.

Acknowledgments

Titanic

- K. Felking, H.P. Leighly, and A. Jankovic, *J. Metals*, (1998).
- T. Foecke, *National Institute of Standards and Technology*, NIST Report NIST-IR 6118.
- R.A. Souza, “The Titanic Disaster: Two Theories,” *Materials Research Society Web Site*.
- D.L. Johnson, W.N. Weins, J.D. Makinson, and D.A. Martinez, in **Microstructural Science**, ASM, 2000.
- Titanic Historical Society

World Trade Center

- T.W. Eagar and C. Musso, *JOM*, (2001) December 8-11.
- J.R. Barnett, R.R. Biederman, and R.D. Sisson, *JOM*, (2001) December 18.

Columbia Space Shuttle

- New York Times
- NASA
- Columbia Accident Investigation Board Final Report, 2003
- D.W. McKee, *Carbon*, (1986) **24** 737.
- K.K. Chawla, **Ceramic Matrix Composites**, Kluwer Academic, (2003).